

STATISTICAL LINEARIZATION OF NONLINEAR STRUCTURAL SYSTEMS WITH SINGULAR MATRICES

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ABSTRACT

A generalized statistical linearization technique is developed for determining approximately the stochastic response of nonlinear dynamic systems with singular matrices. This system modeling can arise when a greater than the minimum number of coordinates is utilized, and can be advantageous, for instance, in cases of complex multibody systems where the explicit formulation of the equations of motion can be a nontrivial task. In such cases, the introduction of additional/redundant degrees of freedom can facilitate the formulation of the equations of motion in a less labor intensive manner. Specifically, relying on the generalized matrix inverse theory and on the Moore-Penrose (M-P) matrix inverse, a family of optimal and response dependent equivalent linear matrices is derived. This set of equations in conjunction with a generalized excitation-response relationship for linear systems leads to an iterative determination of the system response mean vector and covariance matrix. Further, it is proved that setting the arbitrary element in the M-P solution for the equivalent linear matrices equal to zero yields a mean square error at least as low as the error corresponding to any non-zero value of the arbitrary element. This proof greatly facilitates the practical implementation of the technique as it promotes the utilization of the intuitively simplest solution among a family of possible solutions. A pertinent numerical example demonstrates the validity of the generalized technique.

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22 Moore-Penrose Inverse.

23 INTRODUCTION

24 The dynamic analysis of nonlinear systems subjected to stochastic excitation has received con-
25 siderable attention over the last six decades; see Lin (1967), Newland (1993), Grigoriu (2002),
26 and Li and Chen (2009) for some indicative books, as well as Naess and Johnsen (1993), Pirrotta
27 and Santoro (2011), Kougioumtzoglou and Spanos (2014) and Kougioumtzoglou et al. (2015) for
28 some recently developed techniques such as the ones based on path integrals.

29 Undoubtedly, a critical role in the study of linear and nonlinear structural dynamic systems
30 plays the procedure that is followed for the formulation of the system equations of motion, and
31 in particular, the number of coordinates that are utilized. In general, using the minimum number
32 of coordinates (generalized coordinates) for formulating the system equations of motion yields
33 matrices that are not only non-singular, but also symmetric and positive definite (e.g. Pars 1979,
34 Roberts and Spanos 2003). This feature of "well-behaved" matrices greatly facilitates the analysis
35 of such dynamic systems since a number of techniques exist for determining efficiently the system
36 response statistics (e.g. Roberts and Spanos 2003).

37 Nevertheless, it can be argued that there are cases, especially for large scale multi-body sys-
38 tems, where utilizing generalized coordinates for the system modeling is not always the most
39 efficient approach. Specifically, the complexity of the equations of motion (and thus, the effort
40 for their formulation) may increase rapidly with increasing the number of constituent bodies (e.g.
41 Pradhan et al. 1997, Nikravesh et al. 1985, Schiehlen 1984, Schutte and Udwadia 2011, Mariti
42 et al. 2011). In fact, in many cases the choice of modeling utilizing the minimum number of
43 degrees-of-freedom (DOFs)/coordinates relates to excessive computational cost (e.g. Featherstone
44 1987, Bae and Haug 1987, Critchley and Anderson 2003, de Falco et al. 2009). On the other hand,
45 employing additional/redundant, not independent, coordinates in the structural system modeling
46 yields, typically, equations with singular mass, damping and stiffness matrices (e.g. Laulusa and
47 Bauchau 2007, Nikravesh et al. 1985, Udwadia and Wanichanon 2013). Note in passing that uti-

48 lizing redundant coordinates is not the only reason for the appearance of singular matrices in the
49 system equations of motion. For instance, singularities may arise in certain applications such as in
50 the rotational motion of rigid bodies even if generalized coordinates are employed (Udwadia and
51 Wanichanon 2013, Nikravesh et al. 1985, Udwadia and Schutte 2010). Further, besides the case
52 where theoretically non-singular, but numerically ill-conditioned matrices may also appear (e.g.
53 Kawano et al. 2013), singular matrices are naturally met in the formulation of the equations of mo-
54 tion of a certain class of smart structures. In this class of vibrating systems, the system mechanical
55 equation of motion is coupled with the electrical equation yielding a differential-algebraic system
56 of equations with a singular mass matrix (e.g. Xu and Koko 2004, Kawano et al. 2013, Kamada
57 et al. 1997). Note that systems described by a set of differential-algebraic equations belong to
58 the wider class of the so-called descriptor systems (e.g. Kalogeropoulos et al. 2014, Gashi and
59 Pantelous 2015).

60 Although it can be argued that in many cases (in particular when relatively complex systems
61 are considered) the latter "unconventional" modeling can be advantageous from a computational
62 efficiency perspective (e.g. Udwadia and Kalaba 2007, Mariti et al. 2011), standard solution
63 techniques (e.g. a state-variable formulation in conjunction with a complex modal analysis), that
64 inherently assume the existence of non-singular matrices, cannot be applied in a straightforward
65 manner. To address this challenge, the authors recently developed a solution framework for deter-
66 mining the stochastic response of linear dynamic systems with singular matrices (Fragkoulis et al.
67 2016).

68 In this paper, the aforementioned solution framework is generalized to account for nonlinear
69 systems. Specifically, the popular and versatile statistical linearization approximate methodology
70 (e.g. Roberts and Spanos 2003) is generalized herein to account for systems with singular matrices.
71 In this regard, relying on the generalized matrix inverse theory and on the Moore-Penrose (M-P)
72 matrix inverse, a family of optimal and response dependent equivalent linear matrices is derived.
73 This set of equations in conjunction with a recently derived (e.g. Fragkoulis et al. 2016) linear sys-
74 tem generalized excitation-response relationship leads to an iterative determination of the system

75 response mean vector and covariance matrix. Further, it is proved that setting the arbitrary element
 76 in the M-P solution for the equivalent linear matrices equal to zero yields a mean square error at
 77 least as low as the error corresponding to any non-zero value of the arbitrary element. A pertinent
 78 numerical example demonstrates the validity of the generalized technique.

79 MOORE-PENROSE THEORY ELEMENTS

80 In this section, elements of the generalized matrix inverse theory, and in particular of the
 81 Moore-Penrose (M-P) inverse, are provided for completeness.

82 **Definition.** *If $\mathbf{A} \in \mathbb{C}^{m \times n}$ then \mathbf{A}^+ is the unique matrix in $\mathbb{C}^{n \times m}$ so that*

$$83 \quad \mathbf{A}\mathbf{A}^+\mathbf{A} = \mathbf{A} \quad , \quad \mathbf{A}^+\mathbf{A}\mathbf{A}^+ = \mathbf{A}^+,$$

$$84 \quad (\mathbf{A}\mathbf{A}^+)^* = \mathbf{A}\mathbf{A}^+ \quad , \quad (\mathbf{A}^+\mathbf{A})^* = \mathbf{A}^+\mathbf{A}. \quad (1)$$

85 The matrix \mathbf{A}^+ is known as the M-P inverse of \mathbf{A} . The M-P inverse of a square matrix exists
 86 for any arbitrary $\mathbf{A} \in \mathbb{C}^{n \times n}$, and if \mathbf{A} is non-singular, \mathbf{A}^+ coincides with \mathbf{A}^{-1} . Eq. (1) represents
 87 the so-called M-P equations. Further, the M-P inverse of any $m \times n$ matrix \mathbf{A} can be determined
 88 by employing various techniques and methodologies, such as a number of recursive formulae (e.g.,
 89 Campbell and Meyer 1979, Greville 1960), and provides a tool for solving equations of the form

$$90 \quad \mathbf{A}\mathbf{x} = \mathbf{b}, \quad (2)$$

91 where \mathbf{A} is a rectangular $m \times n$ matrix, \mathbf{x} is an n vector and \mathbf{b} is an m vector. For a singular square
 92 matrix \mathbf{A} , i.e. $\det \mathbf{A} \neq 0$, utilizing the M-P inverse, Eq. (2) yields

$$93 \quad \mathbf{x} = \mathbf{A}^+\mathbf{b} + (\mathbf{I} - \mathbf{A}^+\mathbf{A})\mathbf{y}, \quad (3)$$

94 where \mathbf{y} is an arbitrary n vector. A more detailed presentation of the topic can be found in Ben-
 95 Israel and Greville (2003) and Campbell and Meyer (1979).

EQUATIONS OF MOTION OF A NONLINEAR MDOF SYSTEM WITH SINGULAR MATRICES

The equations of motion of a general nonlinear n degree-of-freedom (n -DOF) system are given by

$$M\ddot{q} + C\dot{q} + Kq + \Phi(q, \dot{q}, \ddot{q}) = Q(t), \quad (4)$$

where M , C and K are the $n \times n$ mass, damping and stiffness matrices, respectively. Further, q is the n vector containing the n (generalized) displacements of the system and Q is the n vector containing the n (generalized) forces, corresponding to q . Finally, $\Phi(q, \dot{q}, \ddot{q})$ is a nonlinear n vector of the (generalized) coordinates vector q and its derivatives. Considering next an alternative formulation of the equations of motion, where more than the minimum number coordinates are employed (e.g. Udwadia and Schutte 2010; Fragkoulis et al. 2016), Eq. (4) can take the form

$$M_x\ddot{x} + C_x\dot{x} + K_x x + \Phi_x(x, \dot{x}, \ddot{x}) = Q_x(t), \quad (5)$$

where x stands for an l -DOF vector of coordinates ($l \geq n$), Q_x is the l vector of the external forces and M_x , C_x and K_x are the $l \times l$ mass, damping and stiffness matrices, respectively. The augmented nonlinear vector for the l -DOF system is given by $\Phi_x(x, \dot{x}, \ddot{x})$. Further, a number of constraint equations of the form

$$A(x, \dot{x}, t)\ddot{x} = b(x, \dot{x}, t), \quad (6)$$

may arise that practically enforce the connection relations between the considered subsystems (e.g. Udwadia and Phohomsiri 2006). These arising constraints yield in turn a number of additional forces, and thus, Eq. (5) becomes

$$M_x\ddot{x} + C_x\dot{x} + K_x x + \Phi_x(x, \dot{x}, \ddot{x}) = Q_x(t) + Q_x^c(t), \quad (7)$$

where $Q_x^c(t)$ are the additional aforementioned forces. Also, the presence of constraints yields

118 virtual displacements described by the l vector w , which is any non-zero vector satisfying the
 119 condition

$$120 \quad \mathbf{A}w = \mathbf{0}, \quad (8)$$

121 and at any instant of time t can be expressed as

$$122 \quad w^T \mathbf{Q}_x^c = w^T \mathbf{N}. \quad (9)$$

123 The l vector \mathbf{N} describes the nature of the non-ideal constraints and can be obtained by experi-
 124 mentation and/or observation. Taking into consideration Eq. (3), the solution to Eq. (8) becomes

$$125 \quad w = (\mathbf{I} - \mathbf{A}^+ \mathbf{A})y, \quad (10)$$

126 or, equivalently,

$$127 \quad w = \tilde{\mathbf{A}}y, \quad (11)$$

128 where

$$129 \quad \tilde{\mathbf{A}} = \mathbf{I} - \mathbf{A}^+ \mathbf{A}, \quad (12)$$

130 and y is an arbitrary l vector; therefore, Eq. (9) takes the form

$$131 \quad \tilde{\mathbf{A}}\mathbf{Q}_x^c = \tilde{\mathbf{A}}\mathbf{N}. \quad (13)$$

132 Next, multiplying Eq. (7) by $\tilde{\mathbf{A}}$ and considering Eq. (13) yields

$$133 \quad \tilde{\mathbf{A}}\{M_x \ddot{x} + C_x \dot{x} + K_x x + \Phi_x\} = \tilde{\mathbf{A}}(\mathbf{Q}_x + \mathbf{N}). \quad (14)$$

134 Further, without loss of generality and for facilitating the ensuing analysis, the m vector b in Eq.
 135 (6) is assumed to be of the form

$$136 \quad b = F - E\dot{x} - Lx. \quad (15)$$

137 Considering next Eqs. (6), (14) and (15) yields

$$138 \quad \bar{M}_x \ddot{x} = \begin{bmatrix} \tilde{A}(Q_x + N) \\ F \end{bmatrix} - \begin{bmatrix} \tilde{A}C_x \dot{x} \\ E \dot{x} \end{bmatrix} - \begin{bmatrix} \tilde{A}K_x x \\ Lx \end{bmatrix} - \begin{bmatrix} \tilde{A}\Phi_x \\ 0 \end{bmatrix}, \quad (16)$$

139 or, equivalently,

$$140 \quad \bar{M}_x \ddot{x} = \begin{bmatrix} \tilde{A}(Q_x + N + S) \\ b \end{bmatrix}. \quad (17)$$

141 In Eq. (17), the $(m + l) \times l$ matrix \bar{M}_x and the l vector S are given by

$$142 \quad \bar{M}_x = \begin{bmatrix} \tilde{A}M_x \\ A \end{bmatrix}, \quad (18)$$

143 and

$$144 \quad S = -\Phi_x - C_x \dot{x} - K_x x, \quad (19)$$

145 respectively. Considering the M-P inverse, \bar{M}_x^+ , of the $(m + l) \times l$ matrix \bar{M}_x , the solution to Eq.
146 (17) is given by

$$147 \quad \ddot{x} = \bar{M}_x^+ \begin{bmatrix} \tilde{A}(Q_x + N + S) \\ b \end{bmatrix} + (I - \bar{M}_x^+ \bar{M}_x)y. \quad (20)$$

148 Further, according to *Lemma 4* in Udwardia and Shutte (2010), the relationship

$$149 \quad \bar{M}_x^+ \begin{bmatrix} (Q_x + A^+ z) + N + S \\ b \end{bmatrix} = \bar{M}_x^+ \begin{bmatrix} Q_x + N + S \\ b \end{bmatrix}, \quad (21)$$

150 where \bar{M}_x is the matrix defined in Eq. (18), holds true for any l vector z . Therefore, by setting
151 $z = -A(Q_x + N + S)$, Eq. (21) yields

$$152 \quad \bar{M}_x^+ \begin{bmatrix} \tilde{A}(Q_x + N + S) \\ b \end{bmatrix} = \bar{M}_x^+ \begin{bmatrix} Q_x + N + S \\ b \end{bmatrix}. \quad (22)$$

153 Taking into consideration Eq. (22), Eq. (20) degenerates to the form

$$154 \quad \ddot{\mathbf{x}} = \bar{\mathbf{M}}_x^+ \begin{bmatrix} \mathbf{Q}_x + \mathbf{N} + \mathbf{S} \\ \mathbf{b} \end{bmatrix} + (\mathbf{I} - \bar{\mathbf{M}}_x^+ \bar{\mathbf{M}}_x) \mathbf{y}; \quad (23)$$

155 whereas considering ideal constraints, i.e. $\mathbf{N} = \mathbf{0}$, Eq. (23) becomes

$$156 \quad \ddot{\mathbf{x}} = \bar{\mathbf{M}}_x^+ \begin{bmatrix} \mathbf{Q}_x + \mathbf{S} \\ \mathbf{b} \end{bmatrix} + (\mathbf{I} - \bar{\mathbf{M}}_x^+ \bar{\mathbf{M}}_x) \mathbf{y}. \quad (24)$$

157 Taking into account Eqs. (19) and (24), the acceleration vector of the system takes the form

$$158 \quad \ddot{\mathbf{x}} = \bar{\mathbf{M}}_x^+ \left\{ -\tilde{\mathbf{C}}_x \dot{\mathbf{x}} - \tilde{\mathbf{K}}_x \mathbf{x} - \tilde{\mathbf{\Phi}}_x + \tilde{\mathbf{Q}}_x \right\} + (\mathbf{I} - \bar{\mathbf{M}}_x^+ \bar{\mathbf{M}}_x) \mathbf{y}, \quad (25)$$

159 where the $(m+l) \times l$ matrices $\tilde{\mathbf{C}}_x$, $\tilde{\mathbf{K}}_x$, as well as the $(m+l)$ vector $\tilde{\mathbf{Q}}_x$ are given by

$$160 \quad \tilde{\mathbf{C}}_x = \begin{bmatrix} \mathbf{C}_x \\ \mathbf{E} \end{bmatrix}, \quad (26)$$

$$161 \quad \tilde{\mathbf{K}}_x = \begin{bmatrix} \mathbf{K}_x \\ \mathbf{L} \end{bmatrix}, \quad (27)$$

163 and

$$164 \quad \tilde{\mathbf{Q}}_x = \begin{bmatrix} \mathbf{Q}_x \\ \mathbf{F} \end{bmatrix}, \quad (28)$$

165 respectively. Finally, the $(m+l)$ nonlinear vector $\tilde{\mathbf{\Phi}}_x$ is given by

$$166 \quad \tilde{\mathbf{\Phi}}_x = \begin{bmatrix} \mathbf{\Phi}_x \\ \mathbf{0} \end{bmatrix}. \quad (29)$$

167 It is noted that the simplified expression for the response acceleration, Eq. (25), facilitates signifi-

168 cantly (e.g. Fragkoulis et al. 2016) an efficient state variable formulation of the original equations
 169 of motion. Overall, the augmented system of equations can be concisely written in the alternative
 170 form

$$171 \quad \bar{M}_x \ddot{x} + \bar{C}_x \dot{x} + \bar{K}_x x + \bar{\Phi}_x(x, \dot{x}, \ddot{x}) = \bar{Q}_x(t) \quad (30)$$

172 where \bar{M}_x , \bar{C}_x and \bar{K}_x denote the $(m + l) \times l$ augmented mass, damping and stiffness matrices,
 173 $\bar{\Phi}_x(x, \dot{x}, \ddot{x})$ is the $(m + l)$ nonlinear vector of the system and \bar{Q}_x denotes the $(m + l)$ augmented
 174 excitation vector. The augmented mass matrix is given by Eq. (18), whereas the augmented damp-
 175 ing and stiffness matrices are given by

$$176 \quad \bar{C}_x = \begin{bmatrix} \tilde{A}C_x \\ E \end{bmatrix} \quad (31)$$

177 and

$$178 \quad \bar{K}_x = \begin{bmatrix} \tilde{A}K_x \\ L \end{bmatrix}, \quad (32)$$

179 respectively. Finally, the $(m + l)$ vector \bar{Q}_x as well as the $(m + l)$ nonlinear vector $\bar{\Phi}_x$ are given
 180 by

$$181 \quad \bar{Q}_x = \begin{bmatrix} \tilde{A}Q_x \\ F \end{bmatrix} \quad (33)$$

182 and

$$183 \quad \bar{\Phi}_x = \begin{bmatrix} \tilde{A}\Phi_x \\ 0 \end{bmatrix}. \quad (34)$$

184 **A GENERALIZED STATISTICAL LINEARIZATION METHODOLOGY**

185 Statistical linearization has been one of the most versatile approximate methodologies for de-
 186 termining the stochastic response of nonlinear systems efficiently (e.g. Roberts and Spanos 2003).
 187 The main objective of the methodology relates to the replacement of the original nonlinear system
 188 with an "equivalent linear" one by appropriately minimizing the error vector corresponding to the

189 difference between the two systems. The rationale behind this procedure is that closed form analyt-
190 ical expressions exist for the response statistics of a linear system, and thus, the response statistics
191 of the equivalent linear system can be used as an approximation for the response statistics of the
192 original nonlinear system. According to the standard implementation of the methodology, the min-
193 imization criterion relates typically to the mean square error, whereas the Gaussian assumption for
194 the system response probability density functions (PDFs) is commonly adopted (e.g. Roberts and
195 Spanos 2003). Note, that although more sophisticated implementations of the statistical lineariza-
196 tion that relax the aforementioned assumptions and/or employ various other minimization criteria
197 (e.g. Socha 2008) exist, these versions typically lack versatility. In this regard, one of the reasons
198 for the wide utilization of the standard statistical linearization methodology has been, undoubt-
199 edly, its versatility in addressing a wide range of nonlinear behaviors. In particular, the Gaussian
200 response assumption in conjunction with the mean square error minimization criterion facilitates
201 the derivation of closed form expressions for the equivalent linear elements (e.g. stiffness, damping
202 coefficients, etc) as functions of the response statistics. Further, regarding the stochastic response
203 determination of linear systems, the authors have recently generalized certain concepts and solu-
204 tion techniques of the standard random vibration theory (e.g. Roberts and Spanos 2003, Li and
205 Chen 2009) to account for systems with singular matrices (see Fragkoulis et al. 2016). These
206 generalized techniques are utilized in the ensuing analysis for developing a generalized statistical
207 linearization methodology.

208 Next, the statistical linearization approximate methodology is generalized to account for the
209 nonlinear system with singular matrices of Eq. (30). To this aim, following closely Roberts and
210 Spanos (2003), an equivalent linear system is sought in the form

$$211 \quad (\bar{M}_x + \bar{M}_e)\ddot{x} + (\bar{C}_x + \bar{C}_e)\dot{x} + (\bar{K}_x + \bar{K}_e)x = \bar{Q}_x(t), \quad (35)$$

212 where \bar{M}_e , \bar{C}_e and \bar{K}_e denote the equivalent linear $(m + l) \times l$ mass, damping and stiffness
213 matrices, respectively, to account for the nonlinearity of the original system.

214 The error vector, $\boldsymbol{\varepsilon}$, between the nonlinear and the equivalent linear system is defined as

$$215 \quad \boldsymbol{\varepsilon} = \bar{\boldsymbol{\Phi}}_{\boldsymbol{x}}(\boldsymbol{x}, \dot{\boldsymbol{x}}, \ddot{\boldsymbol{x}}) - \bar{\boldsymbol{M}}_e \ddot{\boldsymbol{x}} - \bar{\boldsymbol{C}}_e \dot{\boldsymbol{x}} - \bar{\boldsymbol{K}}_e \boldsymbol{x}. \quad (36)$$

216 Further, the mean square error is chosen to be minimized (e.g. Roberts and Spanos 2003), i.e.,

$$217 \quad \mathbb{E}[\boldsymbol{\varepsilon}^T \boldsymbol{\varepsilon}] = \text{minimum}, \quad (37)$$

218 for determining the equivalent linear matrices. This yields the equations

$$219 \quad \frac{\partial}{\partial m_{ij}} \mathbb{E}[\boldsymbol{\varepsilon}^T \boldsymbol{\varepsilon}] = 0, \quad (38)$$

$$220 \quad \frac{\partial}{\partial c_{ij}} \mathbb{E}[\boldsymbol{\varepsilon}^T \boldsymbol{\varepsilon}] = 0 \quad (39)$$

221 and

$$222 \quad \frac{\partial}{\partial k_{ij}} \mathbb{E}[\boldsymbol{\varepsilon}^T \boldsymbol{\varepsilon}] = 0, \quad (40)$$

223 where m_{ij}^e , c_{ij}^e and k_{ij}^e are the (i, j) elements of the matrices $\bar{\boldsymbol{M}}_e$, $\bar{\boldsymbol{C}}_e$ and $\bar{\boldsymbol{K}}_e$, respectively. Fur-

224 thermore, combining Eqs. (36) with (37), the minimization criterion can be equivalently written

225 as

$$226 \quad \sum_{i=1}^{m+l} D_i^2 = \text{minimum}, \quad (41)$$

227 where

$$228 \quad D_i^2 = \mathbb{E} \left\{ \left[\bar{\boldsymbol{\Phi}}_{i,\boldsymbol{x}} - \sum_{j=1}^l (m_{ij}^e \ddot{x}_j + c_{ij}^e \dot{x}_j + k_{ij}^e x_j) \right]^2 \right\}, \quad i = 1, 2, \dots, (m+l) \quad (42)$$

229 and

$$230 \quad \bar{\boldsymbol{\Phi}}_{\boldsymbol{x}} = [\bar{\boldsymbol{\Phi}}_{i,\boldsymbol{x}}(\boldsymbol{x}, \dot{\boldsymbol{x}}, \ddot{\boldsymbol{x}})]^T, \quad i = 1, 2, \dots, (m+l). \quad (43)$$

231 Clearly, due to the form of the expression in Eq. (41), the minimization criterion can be equiva-

232 lently written as

$$233 \quad D_i^2 = \text{minimum}, \quad i = 1, 2, \dots, (m + l). \quad (44)$$

234 Next, minimizing Eq. (44) yields the equations

$$235 \quad \mathbb{E} [\bar{\Phi}_{i,x} \hat{\boldsymbol{x}}] = \mathbb{E} [\hat{\boldsymbol{x}} \hat{\boldsymbol{x}}^T] \begin{bmatrix} \mathbf{k}_{i*}^{eT} \\ \mathbf{c}_{i*}^{eT} \\ \mathbf{m}_{i*}^{eT} \end{bmatrix}, \quad i = 1, 2, \dots, (m + l). \quad (45)$$

236 The $3l$ vector $\hat{\boldsymbol{x}}$ is defined as $\hat{\boldsymbol{x}} = (\boldsymbol{x}, \dot{\boldsymbol{x}}, \ddot{\boldsymbol{x}})^T$ and \mathbf{m}_{i*}^{eT} , \mathbf{c}_{i*}^{eT} and \mathbf{k}_{i*}^{eT} correspond to the i^{th} row of
 237 $\bar{\mathbf{M}}_e$, $\bar{\mathbf{C}}_e$ and $\bar{\mathbf{K}}_e$, respectively. Further, adopting the standard Gaussian response assumption, the
 238 term on the left hand side of Eq. (45) is given by (Roberts and Spanos 2003)

$$239 \quad \mathbb{E} [\bar{\Phi}_{i,x} \hat{\boldsymbol{x}}] = \mathbb{E} [\hat{\boldsymbol{x}} \hat{\boldsymbol{x}}^T] \mathbb{E} [\nabla \bar{\Phi}_x(\hat{\boldsymbol{x}})]. \quad (46)$$

240 Combining next Eqs. (45) with (46) yields

$$241 \quad \mathbb{E} [\hat{\boldsymbol{x}} \hat{\boldsymbol{x}}^T] \begin{bmatrix} \mathbf{k}_{i*}^{eT} \\ \mathbf{c}_{i*}^{eT} \\ \mathbf{m}_{i*}^{eT} \end{bmatrix} = \mathbb{E} [\hat{\boldsymbol{x}} \hat{\boldsymbol{x}}^T] \mathbb{E} \begin{bmatrix} \frac{\partial \bar{\Phi}_{i,x}}{\partial \boldsymbol{x}} \\ \frac{\partial \bar{\Phi}_{i,x}}{\partial \dot{\boldsymbol{x}}} \\ \frac{\partial \bar{\Phi}_{i,x}}{\partial \ddot{\boldsymbol{x}}} \end{bmatrix}, \quad i = 1, 2, \dots, (m + l). \quad (47)$$

242 Clearly, the determination of the equivalent linear elements in Eq. (47) requires the inversion of
 243 $\mathbb{E} [\hat{\boldsymbol{x}} \hat{\boldsymbol{x}}^T]$. Thus, the question arises whether this $3l \times 3l$ matrix $\mathbb{E} [\hat{\boldsymbol{x}} \hat{\boldsymbol{x}}^T]$, which appears in both sides
 244 of Eq. (47), is singular or not. As proved in Spanos and Iwan (1978), a necessary and sufficient
 245 condition for $\mathbb{E} [\hat{\boldsymbol{x}} \hat{\boldsymbol{x}}^T]$ to be singular is that at least one of the components of the $3l$ vector $\hat{\boldsymbol{x}}$, can
 246 be expressed as a linear combination of the remaining components. In this regard, note that in
 247 the formulation herein it is assumed a priori that the elements of the coordinates vector \boldsymbol{x} are not
 248 independent with each other as more than the minimum coordinates are utilized in forming the
 249 equations of motion. Thus, it is anticipated that some of the elements of $\hat{\boldsymbol{x}}$ are linearly dependent.

250 Therefore, the matrix $\mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]$ in Eq. (47) is singular and its M-P inverse, denoted as $\mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]^+$,
 251 is employed next for determining an expression for the elements m_{ij}^e , c_{ij}^e and k_{ij}^e of the equivalent
 252 linear augmented matrices. Considering Eq. (3), Eq. (47) is written in the form

$$253 \begin{bmatrix} \mathbf{k}_{i*}^{eT} \\ \mathbf{c}_{i*}^{eT} \\ \mathbf{m}_{i*}^{eT} \end{bmatrix} = \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]^+ \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T] \mathbb{E} \begin{bmatrix} \frac{\partial \bar{\Phi}_{i,x}}{\partial \mathbf{x}} \\ \frac{\partial \bar{\Phi}_{i,x}}{\partial \dot{\mathbf{x}}} \\ \frac{\partial \bar{\Phi}_{i,x}}{\partial \ddot{\mathbf{x}}} \end{bmatrix} + \mathbf{g}(\mathbf{y}), \quad i = 1, 2, \dots, (m+l), \quad (48)$$

254 where the $3l$ vector

$$255 \mathbf{g}(\mathbf{y}) = (\mathbf{I} - \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]^+ \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T])\mathbf{y}, \quad (49)$$

256 is the arbitrary part of the M-P inverse based general solution of Eq. (47). At this point, it is
 257 deemed important to note that when the minimum number of coordinates, n , is utilized, $\mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]$ is
 258 a non-singular matrix yielding

$$259 \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]^+ = \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]^{-1}. \quad (50)$$

260 In that case, $\hat{\mathbf{x}} = (\mathbf{q}, \dot{\mathbf{q}}, \ddot{\mathbf{q}})^T$ and, therefore, combining Eqs. (49) with (50), Eq. (48) takes the
 261 well-established form

$$262 \begin{bmatrix} \mathbf{k}_{i*}^{eT} \\ \mathbf{c}_{i*}^{eT} \\ \mathbf{m}_{i*}^{eT} \end{bmatrix} = \mathbb{E} \begin{bmatrix} \frac{\partial \Phi_{i,q}}{\partial \mathbf{q}} \\ \frac{\partial \Phi_{i,q}}{\partial \dot{\mathbf{q}}} \\ \frac{\partial \Phi_{i,q}}{\partial \ddot{\mathbf{q}}} \end{bmatrix}, \quad i = 1, 2, \dots, n. \quad (51)$$

263 Specifically, Eq. (51) represents the celebrated expressions for determining the elements of the
 264 equivalent linear mass, damping and stiffness matrices in the standard formulation of the statistical
 265 linearization methodology (e.g. Roberts and Spanos 2003). Nevertheless, when formulating the
 266 system equations of motion by employing additional DOFs, $\mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]$ is singular and the generalized
 267 version of Eq. (51) (i.e. Eq. (48)) needs to be considered. Regarding Eq. (48), it can be readily
 268 seen that a critical step for the practical implementation of the generalized statistical linearization
 269 methodology is the choice of the arbitrary element \mathbf{y} . It is proved in the following Proposition

270 that the solution obtained by setting the arbitrary element \mathbf{y} equal to zero is not only intuitively the
 271 simplest but it is also at least as good (in the sense of minimizing the mean square error, where
 272 the error ε is defined in Eq. (36)) as any other solution obtained by selecting an arbitrary non-zero
 273 value for \mathbf{y} . Specifically, setting $\mathbf{y} = \mathbf{0}$, Eq. (48) becomes

$$274 \begin{bmatrix} \mathbf{k}_{i*}^{eT} \\ \mathbf{c}_{i*}^{eT} \\ \mathbf{m}_{i*}^{eT} \end{bmatrix} = \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]^+ \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T] \mathbb{E} \begin{bmatrix} \frac{\partial \bar{\Phi}_{i,\mathbf{x}}}{\partial \mathbf{x}} \\ \frac{\partial \bar{\Phi}_{i,\mathbf{x}}}{\partial \dot{\mathbf{x}}} \\ \frac{\partial \bar{\Phi}_{i,\mathbf{x}}}{\partial \ddot{\mathbf{x}}} \end{bmatrix}, \quad i = 1, 2, \dots, (m+l). \quad (52)$$

275 Assume next that $(m_{ij}^e, c_{ij}^e, k_{ij}^e)$ is the set of parameters arising from solving Eq. (52) and
 276 corresponding to the equivalent matrices $\bar{\mathbf{M}}_e, \bar{\mathbf{C}}_e$ and $\bar{\mathbf{K}}_e$. Also, selecting an arbitrary vector
 277 $\mathbf{y} = \mathbf{y}_0 \neq \mathbf{0}$ in Eq. (49), a different set of parameters, $(m'_{ij}, c'_{ij}, k'_{ij})$, corresponding to matrices
 278 $\bar{\mathbf{M}}'_e, \bar{\mathbf{C}}'_e, \bar{\mathbf{K}}'_e$, is obtained via Eq. (48); see also Spanos and Iwan (1978).

279 **Proposition.** Let $D_i^2(m_{ij}^e, c_{ij}^e, k_{ij}^e)$ and $D_i^2(m'_{ij}, c'_{ij}, k'_{ij})$ denote the errors as defined in Eq. (42)
 280 corresponding to the parameters values $m_{ij}^e, c_{ij}^e, k_{ij}^e$ and $m'_{ij}, c'_{ij}, k'_{ij}$, respectively. Then,

$$281 D_i^2(m_{ij}^e, c_{ij}^e, k_{ij}^e) \leq D_i^2(m'_{ij}, c'_{ij}, k'_{ij}), \quad (53)$$

282 for $i = 1, 2, \dots, (m+l)$ and $j = 1, 2, \dots, l$.

283 *Proof.* From Eq. (42) it is seen that the quantity $D_i^2(m_{ij}^e, c_{ij}^e, k_{ij}^e)$ is a quadratic polynomial with
 284 respect to the parameters m_{ij}^e, c_{ij}^e and k_{ij}^e . Therefore, its mixed partial derivatives concerning
 285 $m_{ij}^e, c_{ij}^e, k_{ij}^e$ of order higher than two vanish. Taking into account Eq. (48), the two sets of pa-
 286 rameters are connected via the expressions

$$287 m'_{ij} = m_{ij}^e + g_{m,i}(y_0), \quad (54)$$

$$288 c'_{ij} = c_{ij}^e + g_{c,i}(y_0), \quad (55)$$

$$289 k'_{ij} = k_{ij}^e + g_{k,i}(y_0), \quad (56)$$

290 where the terms $g_{m,i}(y_0), g_{c,i}(y_0), g_{k,i}(y_0), i = 1, 2, \dots, m + l$, represent the arbitrary elements as
 291 defined in Eq. (49). Next, considering a Taylor's expansion around $(m_{ij}^e, c_{ij}^e, k_{ij}^e)$, yields

$$292 \quad D_i^2(m_{ij}^e, c_{ij}^e, k_{ij}^e) = D_i^2(m_{ij}^e, c_{ij}^e, k_{ij}^e) + \sum_{j=1}^l \left(\frac{\partial D_i^2}{\partial m_{ij}^e} g_{m,i}(y_0) + \frac{\partial D_i^2}{\partial c_{ij}^e} g_{c,i}(y_0) + \frac{\partial D_i^2}{\partial k_{ij}^e} g_{k,i}(y_0) \right) \\ 293 \quad + \frac{1}{2} \mathbb{E} \left\{ \left[\sum_{j=1}^l (g_{m,i}(y_0) \ddot{x}_j + g_{c,i}(y_0) \dot{x}_j + g_{k,i}(y_0) x_j) \right]^2 \right\}, \quad (57)$$

294 for $i = 1, 2, \dots, m + l$, where the terms $g_{m,i}(y_0), g_{c,i}(y_0)$ and $g_{k,i}(y_0)$ denote the distance between
 295 the two sets of parameters.

296 Also, taking into account Eqs. (38)-(40), the necessary conditions for minimizing Eq. (44) are

$$297 \quad \frac{\partial D_i^2}{\partial m_{ij}^e} = 0, \quad (58)$$

$$298 \quad \frac{\partial D_i^2}{\partial c_{ij}^e} = 0 \quad (59)$$

299 and

$$300 \quad \frac{\partial D_i^2}{\partial k_{ij}^e} = 0. \quad (60)$$

301 Utilizing then Eqs. (58)-(60), the first sum on the right hand side of Eq. (57) is zero and Eq. (57)
 302 takes the form

$$303 \quad D_i^2(m_{ij}^e, c_{ij}^e, k_{ij}^e) = D_i^2(m_{ij}^e, c_{ij}^e, k_{ij}^e) + \frac{1}{2} \mathbb{E} \{ J_i^2 \}, \quad i = 1, 2, \dots, m + l, \quad (61)$$

304 where

$$305 \quad J_i = \sum_{j=1}^l (g_{m,i}(y_0) \ddot{x}_j + g_{c,i}(y_0) \dot{x}_j + g_{k,i}(y_0) x_j). \quad (62)$$

306 Finally, taking into account that $\mathbb{E} \{ J_i^2 \} \geq 0$, for all $i = 1, 2, \dots, m + l$, Eq. (61) proves the
 307 argument stated in Eq. (53). \square

308 Clearly, based on Eq. (53), utilizing Eq. (52) yields equivalent linear elements corresponding

309 to an error that is at least as small (in a mean square sense) as any other obtained by utilizing a
 310 non-zero \mathbf{y} vector in Eq. (48).

311 At this point, it is noted that comparing the standard Eq. (51) with its generalized counter-
 312 part Eq. (52) the equivalent linear matrices in Eq. (52) have typically a more complex structure
 313 than their counterparts in Eq. (51). Specifically, due to the fact that in Eq. (52) the product
 314 $\mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]^+ \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]$ does not yield a unitary matrix, the equivalent linear matrices are anticipated to
 315 have many more non-zero components than in the case of utilizing Eq. (51). This observation is
 316 further highlighted in the numerical example section. Additionally, the determination of the equiv-
 317 alent linear matrices in Eq. (52) requires the knowledge of the response covariance matrix $\mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]$.
 318 Obviously, an additional system of equations is needed that relates the two sets of unknowns, i.e.
 319 the response covariance matrix and the equivalent linear elements. In this regard, focusing on the
 320 linearized system of Eq. (35), generalized excitation-response relationships recently derived by
 321 the authors can be employed. Specifically, the standard state-variable formulation and the complex
 322 modal analysis were generalized for treating systems with singular matrices and for determining
 323 the augmented system response covariance matrix (see Fragkoulis et al. 2016). In the following
 324 subsections, the basic elements of these approaches are included for completeness.

325 **State variable formulation and analysis**

326 Considering the M-P inverse of the $\bar{\mathbf{M}}_x + \bar{\mathbf{M}}_e$ matrix, the augmented equivalent linear system
 327 of Eq. (35) can be cast in the form

$$328 \quad \dot{\mathbf{p}} = \mathbf{G}_x \mathbf{p} + \mathbf{f}_x, \quad (63)$$

329 where $\mathbf{p} = \begin{bmatrix} \mathbf{x} & \dot{\mathbf{x}} \end{bmatrix}^T$; and the $2l \times 2l$ matrix \mathbf{G}_x and the $2l$ vector \mathbf{f}_x , are given by

$$330 \quad \mathbf{G}_x = \begin{bmatrix} \mathbf{0} & \mathbf{I} \\ -(\bar{\mathbf{M}}_x + \bar{\mathbf{M}}_e)^+(\bar{\mathbf{K}}_x + \bar{\mathbf{K}}_e) & -(\bar{\mathbf{M}}_x + \bar{\mathbf{M}}_e)^+(\bar{\mathbf{C}}_x + \bar{\mathbf{C}}_e) \end{bmatrix} \quad (64)$$

331 and

$$332 \quad \mathbf{f}_x = \begin{bmatrix} \mathbf{0} \\ (\bar{\mathbf{M}}_x + \bar{\mathbf{M}}_e)^+ \bar{\mathbf{Q}}_x \end{bmatrix}, \quad (65)$$

333 respectively. Further, for zero initial conditions, i.e. $\mathbf{p}(0) = \mathbf{0}$, the solution of Eq. (63) is given by

$$334 \quad \mathbf{p}(t) = \int_0^t \exp(\mathbf{G}_x \tau) \mathbf{f}_x(t - \tau) d\tau, \quad (66)$$

335 where the $2l \times 2l$ transition matrix $\exp(\mathbf{G}_x t)$ has the block matrix form

$$336 \quad \exp(\mathbf{G}_x t) = \begin{bmatrix} \mathbf{a}_x(t) & \mathbf{b}_x(t) \\ \mathbf{c}_x(t) & \mathbf{d}_x(t) \end{bmatrix}. \quad (67)$$

337 Combining next Eqs. (66)-(67), the response displacement vector \mathbf{x} is given by

$$338 \quad \mathbf{x}(t) = \int_0^t \mathbf{h}_x(\tau) \bar{\mathbf{Q}}_x(t - \tau) d\tau, \quad (68)$$

339 where

$$340 \quad \mathbf{h}_x(t) = \mathbf{b}_x(t) (\bar{\mathbf{M}}_x + \bar{\mathbf{M}}_e)^+ \quad (69)$$

341 can be construed as the uniquely defined "generalized" impulse response matrix.

342 Note that in deriving Eq. (68) arguments for neglecting the arbitrary term associated with the
 343 M-P inverse of the $\bar{\mathbf{M}}_x + \bar{\mathbf{M}}_e$ matrix have been employed. These relate to uniquely defining a
 344 response acceleration vector (see also Eq. (25)) as suggested by experimental observations; see
 345 Udawadia and Phohomsiri (2006) and Fragkoulis et al. (2016) for a detailed discussion.

346 Next, manipulating Eq. (63) and taking expectations yields the equation for the system re-
 347 sponse covariance matrix in the form

$$348 \quad \dot{\mathbf{V}}_x = \mathbf{G}_x \mathbf{V}_x + \mathbf{V}_x \mathbf{G}_x^T + \mathbf{S}_x, \quad (70)$$

349 where

$$350 \quad \mathbf{S}_x = \int_0^t \exp(\mathbf{G}_x(t - \tau)) [\mathbf{w}_{\eta_x}(t, \tau) + \mathbf{w}_{\eta_x}^T(t, \tau)] d\tau; \quad (71)$$

351 and \mathbf{w}_{η_x} denotes the covariance matrix of the vector

$$352 \quad \boldsymbol{\eta}_x = \mathbf{f}_x(t) - \mathbb{E}[\mathbf{f}_x(t)]. \quad (72)$$

353 Further, for the case where the elements of $\boldsymbol{\eta}_x$ are regarded to be stationary white noises, Eq. (70)
354 degenerates to

$$355 \quad \dot{\mathbf{V}}_x = \mathbf{G}_x \mathbf{V}_x + \mathbf{V}_x \mathbf{G}_x^T + \mathbf{D}_x, \quad (73)$$

356 where \mathbf{D}_x is a real, symmetric, non-negative matrix of constants. Focusing next on the system
357 stationary response, i.e. $\dot{\mathbf{V}}_x = 0$, Eq. (73) becomes a Lyapunov equation of the form

$$358 \quad \mathbf{G}_x \mathbf{V}_x + \mathbf{V}_x \mathbf{G}_x^T + \mathbf{D}_x = \mathbf{0} \quad (74)$$

359 that, notably, does not have a unique solution due to the form of the augmented matrix \mathbf{G}_x as
360 highlighted in Fragkoulis et al. (2016). Nevertheless, recasting the Lyapunov equation in a form
361 that utilizes the Kronecker product, it has been shown that a solution for the response covariance
362 matrix can be provided.

363 **Complex modal analysis**

364 Focusing next on a complex modal analysis treatment, due to the form of matrix \mathbf{G}_x , its eigen-
365 vectors that correspond to its zero eigenvalues are linearly dependent. Thus, a standard eigende-
366 composition analysis cannot be performed as is the case for modeling using generalized coordi-
367 nates. In this regard, the singular value decomposition (SVD) method can be applied for matrix
368 \mathbf{G}_x yielding

$$369 \quad \mathbf{G}_x = \mathbf{U} \boldsymbol{\eta}_x \boldsymbol{\Psi}^*, \quad (75)$$

370 where the diagonal $2l \times 2l$ matrix $\boldsymbol{\eta}_x$ is given by

$$371 \quad \boldsymbol{\eta}_x = \text{diag}(\sigma_1, \sigma_2, \dots, \sigma_r, 0, 0, \dots, 0). \quad (76)$$

372 In Eq. (76), $\sigma_j = \sqrt{\lambda_j}$, $j = 1, 2, \dots, 2l$ denote the singular values of \mathbf{G}_x , whereas the $2l \times 2l$ matrices $\mathbf{U} = [\mathbf{u}_1, \mathbf{u}_2, \dots, \mathbf{u}_{2l}]$ and $\boldsymbol{\Psi} = [\boldsymbol{\psi}_1, \boldsymbol{\psi}_2, \dots, \boldsymbol{\psi}_{2l}]$ are unitary. Further, $\boldsymbol{\psi}_j$ is the eigenvector
 373 corresponding to the singular value σ_j ($j = 1, 2, \dots, 2l$) whereas \mathbf{u}_j is equal to $\mathbf{u}_j = \frac{\mathbf{G}_x \boldsymbol{\psi}_j}{\sigma_j}$.
 374

375 Utilizing next the SVD of Eq. (75), Eq. (63) can be alternatively written as

$$376 \quad \dot{\mathbf{z}}_x = \boldsymbol{\Psi}^* \mathbf{U} \boldsymbol{\eta}_x \mathbf{z}_x + \mathbf{g}_x, \quad (77)$$

377 where

$$378 \quad \mathbf{g}_x = \boldsymbol{\Psi}^* \mathbf{f}_x \quad (78)$$

379 and

$$380 \quad \mathbf{p} = \boldsymbol{\Psi} \mathbf{z}_x. \quad (79)$$

381 Thus, Eq. (66) becomes

$$382 \quad \mathbf{z}_x(t) = \int_0^t \mathbf{H}_x(s) \mathbf{g}_x(t-s) ds, \quad (80)$$

383 where

$$384 \quad \mathbf{H}_x(t) = \exp(\boldsymbol{\Psi}^* \mathbf{U} \boldsymbol{\eta}_x t). \quad (81)$$

385 As pointed out in Fragkoulis et al. (2016), a complex modal analysis does not result in uncou-
 386 pling the coupled system of Eq. (77). Specifically, the product $\boldsymbol{\Psi}^* \mathbf{U}$ does not yield a unitary matrix
 387 as in the case of utilizing the minimum number of coordinates, and thus, $\mathbf{H}_x(t)$ is not diagonal.
 388 Nevertheless, relying on a SVD of matrix \mathbf{G}_x facilitates significantly the numerical computation
 389 of the system response statistics. In particular, considering Eq. (80) and manipulating yields the

390 covariance matrix \mathbf{w}_{z_x} of the response vector z_x in the form

$$391 \quad \mathbf{w}_{z_x}(\tau) = \int_0^\infty \int_0^\infty \mathbf{H}_x(s_1) \mathbf{w}_{g_x}(\tau + s_1 - s_2) \mathbf{H}_x^*(s_2) ds_1 ds_2. \quad (82)$$

392 Of course, the relationship $\mathbf{p} = \Psi z_x$ can be used for determining the covariance matrix of the
393 response vector \mathbf{p} in the form

$$394 \quad \mathbf{w}_p(\tau) = \Psi \mathbf{w}_{z_x}(\tau) \Psi^*. \quad (83)$$

395 **Mechanization of the generalized statistical linearization methodology**

396 Clearly, based on a modeling utilizing more than the minimum number degrees-of-freedom
397 Eqs. (52) and (70) (or alternatively Eqs. (52) and (82)-(83) if a complex modal analysis is em-
398 ployed) constitute a coupled nonlinear system of equations to be solved for determining the system
399 response covariance matrix and the equivalent linear elements. This can be solved by utilizing any
400 appropriate standard numerical optimization scheme (e.g. Nocedal and Wright 2006), or even the
401 following simple iterative procedure. Specifically,

- 402 (i) Assume zero initial (starting) values for the equivalent linear matrices \bar{M}_e , \bar{C}_e , and \bar{K}_e .
- 403
- 404 (ii) Determine the system response covariance matrix via Eq. (70) (or alternatively via Eqs.
405 (82)-(83)).
- 406
- 407 (iii) Utilize the system response covariance matrix values calculated in (ii) to determine the
408 equivalent linear elements via Eq. (52).
- 409
- 410 (iv) Repeat steps (ii) and (iii) until convergence.

NUMERICAL EXAMPLE

As a numerical example the system of two rigid masses m_1 and m_2 shown in Figure 1 is considered. It is assumed that the mass m_1 is connected to the ground by a nonlinear spring of the linear-plus-cubic type and by a linear damper with coefficient c_1 . Further, a mass m_2 is connected to m_1 by a linear spring and a linear damper with coefficients k_2 and c_2 , respectively. The applied random force $Q_2(t)$ is modeled as a white-noise process with a correlation function $w_{Q_2}(t) = 2\pi S_0 \delta(t)$, where S_0 is the (constant) power spectrum value of $Q_2(t)$. Finally, q_1, q_2 are the generalized displacements, as shown in Figure 1. Further, utilizing generalized coordinates the equations of motion governing the system in Figure 1 can be written in the matrix form of Eq. (4), where the matrices \mathbf{M} , \mathbf{C} and \mathbf{K} are given by (see also Roberts and Spanos 2003)

$$\mathbf{M} = \begin{bmatrix} m_1 & 0 \\ 0 & m_2 \end{bmatrix}, \quad \mathbf{C} = \begin{bmatrix} (c_1 + c_2) & -c_2 \\ -c_2 & c_2 \end{bmatrix}, \quad \mathbf{K} = \begin{bmatrix} (k_1 + k_2) & -k_2 \\ -k_2 & k_2 \end{bmatrix}; \quad (84)$$

the coordinate vector \mathbf{q} and the excitation vector $\mathbf{Q}(t)$ are given by

$$\mathbf{q} = \begin{bmatrix} q_1 \\ q_2 \end{bmatrix} \quad (85)$$

and

$$\mathbf{Q} = \begin{bmatrix} 0 \\ Q_2(t) \end{bmatrix}, \quad (86)$$

respectively. Finally, the nonlinear function Φ takes the form

$$\Phi(\mathbf{q}, \dot{\mathbf{q}}, \ddot{\mathbf{q}}) = \begin{bmatrix} \varepsilon_1 k_1 q_1^3 \\ 0 \end{bmatrix}. \quad (87)$$

428 Next, taking into account Eqs. (51) and (87) yields the equivalent linear stiffness matrix

$$429 \quad \mathbf{K}_e = \begin{bmatrix} 3\varepsilon_1 k_1 \sigma_{q_1}^2 & 0 \\ 0 & 0 \end{bmatrix}. \quad (88)$$

430 Focusing next on the stationary system response, i.e. $\dot{\mathbf{V}} = \mathbf{0}$, a standard statistical lineariza-
 431 tion procedure in conjunction with a complex modal analysis treatment (e.g. Roberts and Spanos
 432 2003) for the values $m_1 = m_2 = m = 1, c_1 = c_2 = c = 0.1, k_1 = k_2 = k = 1$ and
 433 $S_0 = 10^{-3}$, is applied. Regarding the numerical implementation, convergence based on the crite-
 434 rion $\left| \frac{\mathbf{K}_e^{j+1} - \mathbf{K}_e^j}{\mathbf{K}_e^j} \right| > 10^{-5}$, where the j index denotes the j -th iteration, is satisfied after eight
 435 iterations. The initial value \mathbf{K}_e^0 has been set equal to zero. Further, by applying a complex modal
 436 analysis treatment, the eigenvalues of the system after the last iteration are

$$437 \quad \lambda_1 = -0.1308 - 1.6389i \quad , \quad \lambda_2 = -0.1308 + 1.6389i,$$

$$438 \quad \lambda_3 = -0.0192 - 0.6422i \quad , \quad \lambda_4 = -0.0192 + 0.6422i, \quad (89)$$

439 whereas the corresponding eigenvectors are

$$440 \quad \mathbf{v}_1^T = \begin{bmatrix} -0.0357 - 0.4466i & 0.0188 + 0.2626i & 0.7366 & -0.4328 - 0.0036i \end{bmatrix},$$

$$441 \quad \mathbf{v}_2^T = \begin{bmatrix} -0.0357 + 0.4466i & 0.0188 - 0.2626i & 0.7366 & -0.4328 + 0.0036i \end{bmatrix},$$

$$442 \quad \mathbf{v}_3^T = \begin{bmatrix} -0.4260 - 0.0014i & -0.7255 & 0.0090 - 0.2736i & 0.0139 - 0.4659i \end{bmatrix},$$

$$443 \quad \mathbf{v}_4^T = \begin{bmatrix} -0.4260 + 0.0014i & -0.7255 & 0.0090 + 0.2736i & 0.0139 + 0.4659i \end{bmatrix}. \quad (90)$$

444 Finally, the obtained covariance matrix of the system response is given by

$$445 \quad \mathbf{V} = \begin{bmatrix} 0.0386 & 0.0639 & 0 & -0.0010 \\ 0.0639 & 0.1102 & 0.0010 & 0 \\ 0 & 0.0010 & 0.0178 & 0.0252 \\ -0.0010 & 0 & 0.0252 & 0.0462 \end{bmatrix}. \quad (91)$$

446 Consider next the system of two masses m_1 and m_2 of the above example modeled as a multi-
 447 body one, and consisting of two separate subsystems as shown in Figure 2; see also Fragkoulis
 448 et al. (2016). In this regard, the two subsystems are related based on the constraint $x_2 = x_1 + d$
 449 (where d is the length of mass m_1). The "unconstrained" equations of motion are derived by
 450 treating the three coordinates (\bar{x}_1 , x_2 and \bar{x}_3) as independent with each other. Next, the equation of
 451 motion of the composite system is derived by including the constraint

$$452 \quad x_2 = x_1 + d \quad (92)$$

453 or, equivalently,

$$454 \quad x_2 = \bar{x}_1 + l_{1,0} + d, \quad (93)$$

455 where $l_{1,0}$ is the unstretched length of the spring k_1 . Further, based on a Lagrangian formulation of
 456 the equations of motion, Eq. (5) becomes (Fragkoulis et al. 2016)

$$457 \quad \mathbf{M}_x = \begin{bmatrix} m_1 & 0 & 0 \\ 0 & m_2 & m_2 \\ 0 & m_2 & m_2 \end{bmatrix}, \quad \mathbf{C}_x = \begin{bmatrix} c_1 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & c_2 \end{bmatrix}, \quad \mathbf{K}_x = \begin{bmatrix} k_1 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & k_2 \end{bmatrix} \quad (94)$$

458 and

$$459 \quad \Phi_x(\mathbf{x}, \dot{\mathbf{x}}, \ddot{\mathbf{x}}) = \begin{bmatrix} \varepsilon_1 k_1 \bar{x}_1^3 \\ 0 \\ 0 \end{bmatrix}, \quad \mathbf{Q}_x = \begin{bmatrix} 0 \\ Q_3 \\ Q_3 \end{bmatrix}, \quad \mathbf{x} = \begin{bmatrix} \bar{x}_1 \\ x_2 \\ \bar{x}_3 \end{bmatrix}, \quad (95)$$

460 where the variables \bar{x}_1 and \bar{x}_3 are defined as

$$461 \quad \bar{x}_1 = x_1 - l_{1,0} \quad \text{and} \quad \bar{x}_3 = x_3 - l_{2,0}. \quad (96)$$

462 In Eq. (96), $l_{2,0}$ is the unstretched length of the spring k_2 . Further, differentiating the constraint of
463 Eq. (93), the two sub-systems are subject to, yields

$$464 \quad \begin{bmatrix} 1 & -1 & 0 \end{bmatrix} \begin{bmatrix} \ddot{\bar{x}}_1 \\ \ddot{\bar{x}}_2 \\ \ddot{\bar{x}}_3 \end{bmatrix} = 0. \quad (97)$$

465 Thus, the matrix \mathbf{A} and the vector \mathbf{b} of Eq. (6) take the form

$$466 \quad \mathbf{A} = \begin{bmatrix} 1 & -1 & 0 \end{bmatrix} \quad \text{and} \quad \mathbf{b} = 0. \quad (98)$$

467 Furthermore, utilizing Eqs. (30), (94), (95) and (98), the new augmented equation of motion can
468 be determined. The matrices $\bar{\mathbf{M}}_x$, $\bar{\mathbf{C}}_x$, $\bar{\mathbf{K}}_x$, as well as the vectors $\bar{\mathbf{Q}}_x$ and $\bar{\mathbf{\Phi}}_x$ are given by

$$469 \quad \bar{\mathbf{M}}_x = \begin{bmatrix} 0.5 & 0.5 & 0.5 \\ 0.5 & 0.5 & 0.5 \\ 0 & 1 & 1 \\ 1 & -1 & 0 \end{bmatrix}, \quad \bar{\mathbf{C}}_x = \begin{bmatrix} 0.05 & 0 & 0 \\ 0.05 & 0 & 0 \\ 0 & 0 & 0.1 \\ 0 & 0 & 0 \end{bmatrix}, \quad \bar{\mathbf{K}}_x = \begin{bmatrix} 0.5 & 0 & 0 \\ 0.5 & 0 & 0 \\ 0 & 0 & 1 \\ 0 & 0 & 0 \end{bmatrix} \quad (99)$$

470 and

$$471 \quad \bar{\mathbf{Q}}_x = \begin{bmatrix} 0.5w(t) \\ 0.5w(t) \\ w(t) \\ 0 \end{bmatrix}, \quad \bar{\mathbf{\Phi}}_x = \begin{bmatrix} 0.5\varepsilon_1 k_1 \bar{x}_1^3 \\ 0.5\varepsilon_1 k_1 \bar{x}_1^3 \\ 0 \\ 0 \end{bmatrix}. \quad (100)$$

472 Applying next Eq. (52) for determining the equivalent linear stiffness matrix $\bar{\mathbf{K}}_e$ yields

$$\begin{aligned}
 473 \quad \mathbf{k}_{1*}^{eT} &= \begin{bmatrix} r_{1,1} & r_{1,2} & r_{1,3} \\ r_{2,1} & r_{2,2} & r_{2,3} \\ r_{3,1} & r_{3,2} & r_{3,3} \end{bmatrix} \begin{bmatrix} \frac{3}{2}\varepsilon_1 k_1 \sigma_{\bar{x}_1}^2 \\ 0 \\ 0 \end{bmatrix}, \quad \mathbf{k}_{2*}^{eT} = \begin{bmatrix} r_{1,1} & r_{1,2} & r_{1,3} \\ r_{2,1} & r_{2,2} & r_{2,3} \\ r_{3,1} & r_{3,2} & r_{3,3} \end{bmatrix} \begin{bmatrix} \frac{3}{2}\varepsilon_1 k_1 \sigma_{\bar{x}_1}^2 \\ 0 \\ 0 \end{bmatrix} \\
 474 \quad \mathbf{k}_{3*}^{eT} &= \begin{bmatrix} r_{1,1} & r_{1,2} & r_{1,3} \\ r_{2,1} & r_{2,2} & r_{2,3} \\ r_{3,1} & r_{3,2} & r_{3,3} \end{bmatrix} \mathbb{E} \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} = \mathbf{0}, \quad \mathbf{k}_{4*}^{eT} = \begin{bmatrix} r_{1,1} & r_{1,2} & r_{1,3} \\ r_{2,1} & r_{2,2} & r_{2,3} \\ r_{3,1} & r_{3,2} & r_{3,3} \end{bmatrix} \mathbb{E} \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix} = \mathbf{0}, \quad (101)
 \end{aligned}$$

475 where $r_{i,j}, i, j = 1, 2, \dots, 9$ denotes the element of the matrix $\mathbf{r} = \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T] + \mathbb{E}[\hat{\mathbf{x}}\hat{\mathbf{x}}^T]$ in position
 476 (i, j) . Hence, considering Eq. (101) the equivalent linear matrix $\bar{\mathbf{K}}_e$ can be concisely written as

$$477 \quad \bar{\mathbf{K}}_e = \frac{3}{2}\varepsilon_1 k_1 \sigma_{\bar{x}_1}^2 \begin{bmatrix} \mathbf{r}_{1,1} & \mathbf{r}_{2,1} & \mathbf{r}_{3,1} \\ \mathbf{r}_{1,1} & \mathbf{r}_{2,1} & \mathbf{r}_{3,1} \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix}. \quad (102)$$

478 At this point, comparing Eqs. (88) and (102) it is noted that although the general form of the
 479 equivalent linear stiffness matrices is similar, the equivalent linear matrix of Eq. (102) has more
 480 non-zero elements. Clearly, this is due to the presence of matrix \mathbf{r} which, unlike the generalized
 481 coordinates modeling case, is not unitary. Next, employing Eq. (64), the matrix \mathbf{G}_x takes the form

$$482 \quad \mathbf{G}_x = \begin{bmatrix} \mathbf{0} & \mathbf{I} \\ -\bar{\mathbf{M}}_x^+(\bar{\mathbf{K}}_x + \bar{\mathbf{K}}_e) & -\bar{\mathbf{M}}_x^+\bar{\mathbf{C}}_x \end{bmatrix}, \quad (103)$$

483 where the M-P inverse of \bar{M}_x is found by Eq. (18) to be equal to

$$484 \quad \bar{M}_x^+ = \begin{bmatrix} 1 & 1 & -1 & 0 \\ 1 & 1 & -1 & -1 \\ -1 & -1 & 2 & 1 \end{bmatrix}. \quad (104)$$

485 Further, as in the case of the covariance matrix obtained in Eq. (91) for the 2–DOF system,
486 a complex modal analysis treatment is utilized for deriving the covariance matrix of the system
487 response. Also, to be consistent with the previously obtained result, the convergence criterion and
488 error are the same as those utilized for deriving Eq. (91). In this regard, convergence is reached
489 after eight iterations. Employing Eqs. (75)-(81), the eigenvalues of the matrix $\Psi^* U \eta_x$, where
490 Ψ, U, η_x are defined in Eq. (75), after the last iteration are

$$491 \quad \lambda_1 = -0.1308 - 1.6389i, \quad \lambda_2 = -0.1308 + 1.6389i, \quad \lambda_3 = -0.0192 - 0.6422i,$$
$$492 \quad \lambda_4 = -0.0192 + 0.6422i, \quad \lambda_5 = 0, \quad \lambda_6 = 0, \quad (105)$$

493 whereas the corresponding eigenvectors are

$$\begin{aligned}
 494 \quad \mathbf{v}_1^T &= \begin{bmatrix} -0.0145 - 0.4629i \\ -0.0432 - 0.0020i \\ 0.4009 + 0.0278i \\ 0.7540 \\ 0.0051 - 0.0227i \\ -0.0343 + 0.2281i \end{bmatrix}, \mathbf{v}_2^T = \begin{bmatrix} -0.0145 + 0.4629i \\ -0.0432 + 0.0020i \\ 0.4009 - 0.0278i \\ 0.7540 \\ 0.0051 + 0.0227i \\ -0.0343 - 0.2281i \end{bmatrix}, \mathbf{v}_3^T = \begin{bmatrix} -0.0308 + 0.0028i \\ 0.0006 - 0.4181i \\ -0.0177 - 0.3418i \\ -0.0060 - 0.0025i \\ 0.6740 \\ 0.5027 - 0.0111i \end{bmatrix}, \\
 495 \quad \mathbf{v}_4^T &= \begin{bmatrix} -0.0308 - 0.0028i \\ 0.0006 + 0.4181i \\ -0.0177 + 0.3418i \\ -0.0060 + 0.0025i \\ 0.6740 \\ 0.5027 + 0.0111i \end{bmatrix}, \mathbf{v}_5^T = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 1 \end{bmatrix}, \mathbf{v}_6^T = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 1 \end{bmatrix}. \tag{106}
 \end{aligned}$$

496 After determining the eigenvalues and eigenvectors of the matrix $\Psi^*U\eta_x$, Eq. (82) evaluated
 497 at $\tau = 0$ takes the form

$$498 \quad \mathbf{w}_{z_x}(0) = - \sum_{i=1}^4 \sum_{j=1}^4 \frac{\mathbf{p}_i(\Psi^*D_x\Psi)\mathbf{p}_j^*}{\lambda_i + \bar{\lambda}_j}, \tag{107}$$

499 where $\lambda_i, i = 1, 2, 3, 4$ are given by Eq. (105) and D_x is a real, symmetric, non-negative matrix of
 500 constants given by

$$501 \quad D_x = \begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 2\pi 10^{-3} \end{bmatrix}. \tag{108}$$

502 In Eq. (107), the expressions $\mathbf{p}_i, i = 1, 2, 3, 4$ denote 6×6 matrices defined in terms of the matrix
503 $\Psi^* \mathbf{U} \boldsymbol{\eta}_x$, as well as the eigenvalues calculated in Eq. (105). For example, \mathbf{p}_1 is defined as (see
504 Fragkoulis et al. 2016 for more details)

$$505 \quad \mathbf{p}_1 = \frac{(\Psi^* \mathbf{U} \boldsymbol{\eta}_x - \lambda_2 \mathbf{I})(\Psi^* \mathbf{U} \boldsymbol{\eta}_x - \lambda_3 \mathbf{I})(\Psi^* \mathbf{U} \boldsymbol{\eta}_x - \lambda_4 \mathbf{I})(\Psi^* \mathbf{U} \boldsymbol{\eta}_x)^2}{(\lambda_1 - \lambda_2)(\lambda_1 - \lambda_3)(\lambda_1 - \lambda_4)\lambda_1^2}. \quad (109)$$

506 Finally, employing Eq. (83), the covariance matrix of the system response is given by

$$507 \quad \mathbf{w}_p(0) = \begin{bmatrix} 0.0386 & 0.0386 & 0.0253 & 0 & 0 & -0.0010 \\ 0.0386 & 0.0386 & 0.0253 & 0 & 0 & -0.0010 \\ 0.0253 & 0.0253 & 0.0210 & 0.0010 & 0.0010 & 0 \\ 0 & 0 & 0.0010 & 0.0178 & 0.0178 & 0.0074 \\ 0 & 0 & 0.0010 & 0.0178 & 0.0178 & 0.0074 \\ -0.0010 & -0.0010 & 0 & 0.0074 & 0.0074 & 0.0136 \end{bmatrix}. \quad (110)$$

508 Indicatively, comparing Eqs. (91) and (110), the variance $\mathbb{E}[q_1^2]$ as well as $\mathbb{E}[\dot{q}_1^2]$ obtained in the
509 first example, coincide with the respective ones in the second one, i.e $\mathbb{E}[\bar{x}_1^2]$ and $\mathbb{E}[\dot{\bar{x}}_1^2]$. Further,
510 taking expectations in the equation that connects the two reference systems, that is $\bar{x}_3 = q_2 - q_1$,
511 and utilizing Eq. (91) yields

$$512 \quad \mathbb{E}[\bar{x}_3^2] = \mathbb{E}[q_2^2] + \mathbb{E}[q_1^2] - 2\mathbb{E}[q_1 q_2] = 0.0210 \quad (111)$$

513 and

$$514 \quad \mathbb{E}[\dot{\bar{x}}_3^2] = \mathbb{E}[\dot{q}_2^2] + \mathbb{E}[\dot{q}_1^2] - 2\mathbb{E}[\dot{q}_1 \dot{q}_2] = 0.0136, \quad (112)$$

515 which are indeed in agreement with the corresponding values in Eq. (110). It can be readily
516 verified that the rest of the elements of the matrix given in Eq. (110) are also in agreement with the
517 respective ones of Eq. (91). It is noted that, alternatively, the response covariance matrix \mathbf{V}_x can

518 be obtained by utilizing a state variable formulation in conjunction with the Lyapunov equation of
519 Eq. (74); see Fragkoulis et al. (2016) for more details.

520 **CONCLUSION**

521 In this paper the standard and popular statistical linearization methodology for determining ap-
522 proximately the stochastic response of nonlinear dynamic systems (Roberts and Spanos 2003) has
523 been generalized to account for systems with singular matrices. This kind of modeling can appear
524 for various reasons, among which is the utilization of additional/redundant coordinates. This can
525 be advantageous for cases of complex multi-body systems, for instance, where formulating the
526 equations of motion in terms of the independent/generalized coordinates can be a non-trivial task.

527 Specifically, relying on the generalized matrix inverse theory and on the M-P inverse of a singu-
528 lar matrix, a family of optimal and response dependent equivalent linear matrices has been derived.
529 Next, this set of equations has been combined with a recently developed by the authors generalized
530 linear system input-output (excitation-response) relationship to yield a coupled system of nonlin-
531 ear algebraic equations. This can be readily solved via an iterative procedure for determining the
532 system response mean vector and covariance matrix. A significant additional contribution of the
533 paper relates to the proof that the solution obtained by setting the arbitrary element in the M-P
534 expression for the equivalent linear matrices equal to zero is at least as good (in a mean square
535 error minimization sense) as any other solution corresponding to a non-zero value for the arbitrary
536 element. This proof greatly facilitates the practical implementation of the technique as it promotes
537 the utilization of the intuitively simplest solution among a family of possible solutions. Finally, a
538 2-DOF nonlinear system modeled by utilizing redundant coordinates is employed in the numer-
539 ical examples section to demonstrate the validity of the herein developed generalized statistical
540 linearization methodology.

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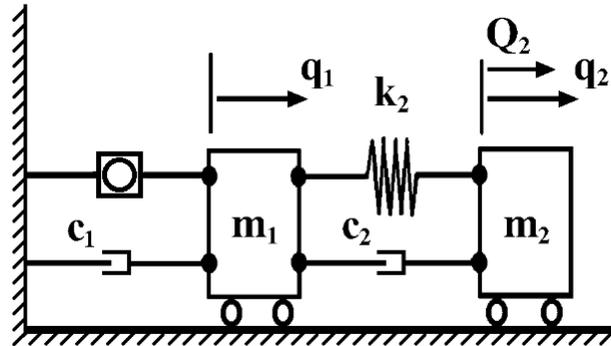


FIG. 1. A two degree-of-freedom nonlinear structural system under stochastic excitation.

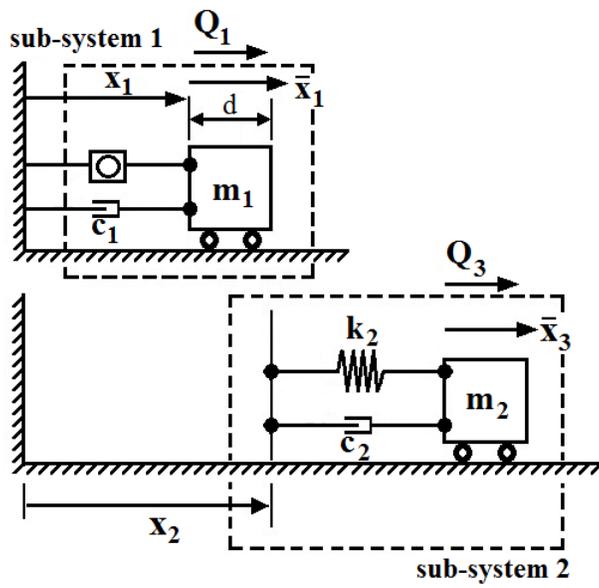


FIG. 2. Modeling of the system shown in Figure 1 using more than two coordinates.