

1 **Local stability of laser-welded stainless steel slender I-sections under** 2 **combined loading**

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14 15 **Abstract**

16 This paper presents a comprehensive experimental and numerical investigation into the local
17 buckling behaviour of laser-welded stainless steel slender I-sections under combined
18 compression and bending moment. A testing programme was firstly carried out, including
19 initial local geometric imperfection measurements and eccentric compression tests on ten
20 laser-welded stainless steel slender I-section stub column specimens. Following the testing
21 programme, a numerical modelling programme was conducted. Finite element models were
22 developed and validated against the test results, and then adopted to perform parametric
23 studies to generate additional numerical data. The obtained test and numerical data were used
24 to evaluate the applicability of the interaction curves in the European and American codes and
25 the continuous strength method to laser-welded stainless steel slender I-sections under
26 combined loading. The evaluation results generally show that the three sets of considered

27 interaction curves offer adequate design accuracy for laser-welded stainless steel slender
28 I-sections under combined compression and major-axis bending moment, but they yield
29 unduly conservative and scattered resistance predictions for laser-welded stainless steel
30 slender I-sections under minor-axis combined loading, owing principally to the conservative
31 minor-axis bending end points. Finally, an improved design interaction curve for the
32 minor-axis combined loading case was developed and shown to yield substantially more
33 accurate and consistent resistance predictions for laser-welded stainless steel slender
34 I-sections under minor-axis combined loading than the three considered design methods. The
35 reliability of the new design interaction curve was confirmed based on a reliability study.

36

37 **Keywords:** Cross-sectional behaviour; Design codes; Eccentric compression tests;
38 Laser-welding; New proposal; Stainless steel slender I-sections.

39

40 **1. Introduction**

41

42 Stainless steel has been increasingly used in building and structural engineering in recent
43 years, owing to its favourable mechanical properties coupled with superior corrosion
44 resistance and durability [1–5], which can greatly reduce the maintenance and inspection
45 work. As an advanced fabrication technique, laser welding can minimise the input heat, which
46 leads to reduced heat affected zones as well as low residual stresses and thermal distortions
47 [6,7]. As a result, laser welding, which is also highly precise, is increasingly adopted for
48 joining stainless steels to form various welded built-up section profiles. Comprehensive

49 research into laser-welded stainless steel structural members with different cross-section
50 profiles subjected to various loading conditions has been conducted, in order to verify their
51 structural responses, examine the codified design rules, and develop improved design methods
52 with good design accuracy and consistency. Previous experimental studies are firstly reviewed
53 herein. Gardner et al. [6] and Ran et al. [8] conducted stub column tests on laser-welded
54 stainless steel I-sections to investigate their cross-section compression resistances and local
55 stability. Bu and Gardner [7] and Theofanous et al. [9] carried out in-plane bending tests on
56 laser-welded stainless steel I-, channel and angle section beams, in order to examine their
57 cross-section bending resistances. Liang et al. [10,11] experimentally investigated the
58 cross-sectional behaviour of laser-welded stainless steel channel sections under combined
59 compression and bending. The global stability of laser-welded stainless steel I- and angle
60 section columns were studied by Gardner et al. [6], Ran et al. [8] and Filipović et al. [12]
61 through a series of concentric compression tests. Bu and Gardner [13] and Kucukler et al. [14]
62 investigated the structural performance of laterally unrestrained laser-welded stainless steel
63 I-section beam-columns under minor-axis and major-axis combined loading, respectively. The
64 literature review revealed that although in-depth research into the structural behaviour and
65 resistances of laser-welded stainless steel components under various loading conditions has
66 been previously conducted, cross-section responses and resistances of laser-welded stainless
67 steel slender I-sections under combined loading remain unexplored, and therefore the present
68 study was initiated.

69

70 In this paper, an experimental programme was firstly carried out, including eccentric

71 compression tests on ten laser-welded stainless steel slender I-section stub columns and
72 supplementary initial local geometric imperfection measurements. The obtained test results
73 were used in a numerical modelling programme for establishing and validating finite element
74 models; upon validation, the numerical models were adopted to perform parametric studies to
75 generate additional numerical data. Based on the obtained test and numerical data, the
76 applicability of the design interaction curves given in EN 1993-1-4 [15], ANSI/AISC 370-21
77 [16] and the continuous strength method [17] to laser-welded stainless steel slender I-sections
78 under combined loading was evaluated. Finally, an improved interaction curve was proposed.

79

80 **2. Experimental Study**

81

82 *2.1. General*

83

84 Given that there have been no experimental results on laser-welded stainless steel I-sections
85 under combined loading, a testing programme was firstly conducted to generate a test data
86 pool. Two different I-sections – I-90×90×3 and I-120×90×3 were adopted in the testing
87 programme, with five stub column specimens prepared for each I-section size. All the ten stub
88 column specimens were fabricated through laser welding from 3 mm thick grade EN 1.4301
89 austenitic stainless steel hot-rolled plates, with the welding techniques and procedures
90 following those given in EN ISO 13919-1 [18]. The two adopted I-sections are classified as
91 Class 4 slender sections based on the slenderness limits set out in EN 1993-1-4 [15] and
92 slender sections according to the cross-section classification framework given in ANSI/AISC

93 370-21 [16]. Overall, the testing programme comprised material testing and residual stress
94 measurements, initial local geometric imperfection measurements and ten eccentric
95 compression tests.

96

97 *2.2. Material testing and membrane residual stress measurements*

98

99 The material stress–strain responses and membrane residual stresses of the studied
100 laser-welded stainless steel I-sections were carefully measured and fully reported by the
101 authors in Ran et al. [8], with the experimental procedures and results briefly summarised
102 herein. Two longitudinal coupons, extracted from the virgin plates with the geometric sizes in
103 compliance with those given in EN ISO 6892-1 [19], were tested in a 300 kN testing machine
104 subjected to displacement control. Fig. 1 shows the material stress–strain curves obtained
105 from the tensile coupon tests, while Table 1 reports the key average measured material
106 properties, including the Young’s modulus E , the 0.2% proof stress f_y , 1.0% proof stress $f_{1.0}$,
107 the ultimate stress f_u , the strain at the ultimate stress ϵ_u , the strain at fracture ϵ_f over a standard
108 gauge length [20] and the coefficients used in the Ramberg–Osgood material model n and $m_{1.0}$
109 [21–24]. The membrane residual stresses in the studied laser-welded stainless steel I-sections
110 were measured by means of the sectioning method, which has been broadly used in previous
111 relevant membrane residual stress measurements [6,25–28], with the detailed procedures and
112 results presented in Ran et al. [8]. The measured membrane residual stresses were found to be
113 well captured by the residual stress predictive model proposed by Gardner et al. [6] for
114 laser-welded stainless steel I-sections, with the pattern shown in Fig. 2 and the distribution

115 parameters (a , b , c and d) presented in Table 2.

116

117 *2.3. Initial local geometric imperfection measurements*

118

119 As an important characteristic of thin-walled steel components, geometric imperfections may
120 affect their structural performance. Initial local geometric imperfection measurements were
121 thus conducted on the ten laser-welded stainless steel I-section stub column specimens. Fig. 3
122 shows the measurement setup, where a stub column specimen is fixed on the moving bench of
123 a milling machine through the use of a vise and a percentage gauge with 0.01 mm precision is
124 tightly mounted onto the machine head to measure the local deviations of each constituent
125 plate element. This measurement setup has been successfully used in previous relevant studies
126 [6–8,10,11,26–29] and was thus also employed in this study. The local imperfection
127 measurements were performed on the cross-sections at mid-height and two third points of the
128 specimen length. For each constituent plate element of the stub column specimen, the initial
129 local geometric imperfections were defined as the deviations from a linear reference line fitted
130 to the corresponding measured data set, while the maximum deviation obtained from the three
131 cross-sections was taken as the initial local geometric imperfection magnitude ω_0 of the stub
132 column specimen, as presented in Table 3. The normalised value ω_0/b_f for each stub column
133 specimen is also given in Table 3, where b_f is the flange width, showing that the largest
134 normalised value ω_0/b_f from all the stub column specimens is 1/257, less than the fabrication
135 tolerance value of 1/100 recommended in EN 1993-1-5 [30].

136

137 *2.4. Eccentric compression tests*

138

139 Eccentric compression tests were carried out on the ten laser-welded stainless steel slender
140 I-section stub column specimens, with six conducted about the major principal axis and four
141 performed about the minor principal axis, in order to investigate their structural performance
142 under combined compression and bending. The geometric dimensions of the ten stub column
143 specimens were measured, as presented in Table 3, including the specimen length L , the outer
144 section height h , the flange width b_f and the plate thickness t . For each stub column specimen,
145 its end sections were then welded with a pair of 6 mm thick flat steel plates. All the eccentric
146 compression tests were conducted in an INSTRON testing machine. The adopted initial
147 loading eccentricities were varied between 5 mm and 80 mm and therefore a wide range of
148 loading combinations were considered in the tests. Pin-ended boundary conditions were
149 applied to each stub column specimen; this was achieved through the use of a pair of
150 knife-edge devices, which were respectively mounted onto the top and bottom ends of the
151 testing machine. As shown in Fig. 4, each knife-edge device comprises a wedge plate with a
152 knife edge and a pit plate with a semi-circular groove. Before testing, each stub column
153 specimen was firstly placed between the top and bottom knife-edge devices, with their
154 relative position adjusted to achieve pre-specified loading eccentricity and proper member
155 alignment. Then, each specimen was fixed to the knife-edge devices through bolting.

156

157 Fig. 4 shows the apparatus used for the eccentric compression tests, including six LVDTs and
158 four strain gauges. The strain gauges were attached to the external surfaces of the two flanges

159 at the specimen mid-height to measure the longitudinal strains at these positions. LVDT-1 to
 160 LVDT-4 were used to measure the rotations at the top and bottom ends, LVDT-5 was adopted
 161 to monitor the end shortening of each specimen, and LVDT-6 was placed at the specimen
 162 mid-height to measure the corresponding lateral deflection. Once the test setup was completed,
 163 a loading rate of 0.2 mm/min was adopted to drive the testing machine to eccentrically
 164 compress each stub column specimen. The longitudinal strains obtained from the two pairs of
 165 strain gauges, together with the mid-height lateral deflections measured from LVDT-6, were
 166 used to determine the actual initial loading eccentricity e_0 of each stub column specimen,
 167 based on Eq. (1) [31–33], where I is the second moment of area about the axis of combined
 168 loading, $\varepsilon_{\max}-\varepsilon_{\min}$ is the longitudinal strain difference, N is the applied compression load and
 169 D is the distance between the two pairs of strain gauges, respectively equal to h and b_f-2d_s for
 170 the major-axis and minor-axis combined loading cases, as graphically depicted in Fig. 5. Note
 171 that Eq. (1) was derived based on the assumption that the structural behaviour was close to
 172 linear elastic, it was suggested that the eccentric compression loads employed for calculating
 173 e_0 be lower than 15% of the estimated failure loads.

$$174 \quad e_0 = \frac{EI(\varepsilon_{\max} - \varepsilon_{\min})}{ND} - \Delta \quad (1)$$

175
 176 The failure modes of the tested laser-welded stainless steel I-section stub column specimens
 177 under minor-axis and major-axis combined loading are displayed in Fig. 6(a) and Fig. 6(b),
 178 respectively, showing significant local buckling coupled with overall flexural deformation
 179 about the axis of combined loading. Fig. 7 presents the full load–end rotation curves for the
 180 two series of tested specimens, while Table 4 presents the key test results, including the actual

181 initial loading eccentricity e_0 , the failure load N_u , the end rotation at the failure load ϕ_u and the
182 corresponding load shortening δ_u and mid-height lateral deflection Δ_u , and the failure moment
183 at the specimen mid-height $M_u=N_u(e_0+\Delta_u)$, Within each specimen series, the stub columns
184 with larger initial loading eccentricities have lower ultimate loads but larger lateral deflections
185 at mid-height, due to the increased effect from bending.

186

187 **3. Numerical Modelling**

188

189 *3.1. General*

190

191 In this section, a numerical modelling programme was carried out by means of the nonlinear
192 finite element (FE) analysis package ABAQUS [34] to supplement the testing programme. FE
193 models on laser-welded stainless steel slender I-section stub column specimens under
194 combined loading were developed and validated against the obtained test results. Upon
195 validation, the developed FE models were employed to perform parametric studies to generate
196 additional numerical data over a wide range of cross-section dimensions and loading
197 combinations.

198

199 *3.2. Development and validation of FE models*

200

201 The ‘S4R’ shell element [34] has been proven to be able to well simulate various stainless
202 steel welded I-section members [7,8,13,14,26,32,33,35–37] and was thus adopted herein.

203 Based on a prior study on the mesh sensitivity, the element length and width were selected as
204 $1.5t$ and t , respectively, which were found to (i) lead to an accurate incorporation of the
205 membrane residual stresses into the FE models and (ii) result in good computational
206 efficiency and accuracy. Regarding the material modelling of the used austenitic stainless
207 steel, the engineering stress–strain response, measured from the material test on Coupon #1
208 (see Fig. 1), was converted into the true stress–strain response and then inputted into
209 ABAQUS [34]. The membrane residual stresses in each laser-welded stainless steel I-section
210 stub column were firstly derived from the predictive model, as depicted in Fig. 2 and Table 2,
211 and then incorporated into the corresponding FE model using the ‘*INITIAL CONDITIONS’
212 command [34]. Fig. 8 shows the membrane residual stresses incorporated into the numerically
213 modelled I-90×90×3 specimens. Upon successful incorporation of the material response and
214 membrane residual stresses, the experimentally adopted pin-ended boundary and eccentric
215 loading conditions were carefully modelled in the development of the FE models. Specifically,
216 for each FE model, its top and bottom end sections were respectively coupled to one reference
217 point that was located (i) longitudinally at a distance of 80 mm away from the corresponding
218 end section and (ii) eccentrically to the combined loading axis with the eccentricity given as
219 the corresponding actual eccentricity e_0 . The top reference point can rotate about the
220 combined loading axis and translate longitudinally, while the bottom reference point only has
221 the rotation about the same axis. The initial local geometric imperfections were also included
222 into each FE model, with the distribution pattern taken as the lowest elastic local buckling
223 mode shape, as derived from a prior elastic eigenvalue buckling analysis [34]. Two local
224 imperfection magnitudes, including the measured value ω_0 and a generalised value $b_f/300$,

225 were adopted to factor the obtained local imperfection distribution profiles.

226

227 Once the FE models were developed, each of them was analysed through the materially and
228 geometrically nonlinear ‘Static, Riks’ analysis [34], in order to derive the numerical results,
229 including the numerical failure load, load–end rotation curve and failure mode. The accuracy
230 of the developed FE models was evaluated through quantitative and graphical comparisons
231 between the derived numerical results and their experimental counterparts. The quantitative
232 comparison results were presented in Table 5, where the FE-to-test failure loads for the stub
233 column specimens are reported, showing that all the two adopted local imperfection
234 magnitudes led to accurate predictions of experimental failure loads; this reflected the
235 insensitivity of the developed FE models to initial local geometric imperfection magnitudes.
236 Graphical comparisons between the test and FE load–end rotation curves for the two series of
237 stub column specimens are presented in Fig. 7, where the experimental responses are
238 accurately captured by their FE counterparts. Moreover, the experimentally obtained failure
239 modes were found to be well simulated by the developed FE models, as shown in Fig. 9(a)
240 and Fig. 9(b) for two typical specimens 90-MI2 and 90-MA3. Upon the quantitative and
241 graphical comparisons, the developed FE models have been shown to be capable of
242 simulating the structural performance of laser-welded stainless steel slender I-sections under
243 combined loading and were thus considered to be validated.

244

245

246

247 3.3. Parametric studies

248

249 Upon validation, the developed FE models were adopted to carry out parametric studies to
250 expand the test data pool over a wide range of cross-section dimensions and loading
251 combinations, beyond those considered in the testing programme. Regarding the modelled
252 I-sections, their cross-section dimensions were carefully selected, ensuring that they were
253 classified as slender I-sections by both EN 1993-1-4 [15] and ANSI/AISC 370-21 [16]. Table
254 6 reports the cross-section dimensions and loading eccentricities used in the parametric
255 studies. Specifically, both the outer section height h and flange width b_f were varied from 60
256 mm to 150 mm, with the plate thickness t varied between 2 mm and 3 mm, leading to a broad
257 spectrum of cross-section dimensions and aspect ratios being examined. The cross-sectional
258 slendernesses [17] were varied from 0.72 to 1.65. It is worth noting that the maximum
259 cross-sectional slenderness of the specimens is 0.80; therefore, the numerical results for
260 I-sections with slenderness greater than 0.80 may require a validation in the future based on
261 further eccentric compression tests. The model length was equal to $3h$ [25]. Moreover, the
262 initial loading eccentricities were varied between 1 mm and 100 mm, leading to a wide
263 spectrum of loading combinations being considered. All the FE models were developed
264 through employing the modelling assumptions, techniques and procedures detailed in Section
265 3.2, but their local imperfection magnitudes were kept at $b_f/300$. The engineering stress–strain
266 response measured from the material test on Coupon #1 (see Fig. 1) was converted into the
267 true stress–strain response and then used in the parametric studies. In total, 400 numerical
268 data on laser-welded stainless steel slender I-sections under combined loading were derived

269 through the parametric studies, with 200 for each principal axis.

270

271

272

273 **4. Design Analysis**

274

275 *4.1. General*

276

277 In this section, the applicability of the design interaction curves given in EN 1993-1-4 [15],
278 AISC/AISC 370-21 [16] and the continuous strength method [17] to laser-welded stainless
279 steel slender I-sections under combined compression and bending was evaluated by
280 comparing the test and FE failure loads N_u against the unfactored design failure loads $N_{u,pred}$.
281 A new design method was also proposed for laser-welded stainless steel slender I-sections
282 under minor-axis combined loading, underpinned by the test and FE data. Table 7 presents the
283 quantitative evaluation of the considered and proposed design methods, where the mean test
284 and numerical to predicted failure load ratios $N_u/N_{u,pred}$ and the corresponding coefficients of
285 variance (COVs) are reported, while Figs 10–13 show the graphical evaluation results, where
286 the test and numerical data are plotted against the considered and proposed design interaction
287 curves, respectively.

288

289 *4.2. EN 1993-1-4 (EC3)*

290

291 The current Eurocode EN 1993-1-4 [15] adopts similar design rules for stainless steel
 292 structures as those given in EN 1993-1-1 [38] for mild steel structures. Regarding Class 4
 293 slender I-sections under combined loading, the Eurocodes [15,38] use a linear interaction
 294 curve, as given by Eq. (2), where $N_{u,pred}$ is the design failure load and $M_{u,y}$ (or
 295 $M_{u,z}=N_{u,pred}(e_0+A_u)$ is the design major-axis (or minor-axis) failure moment, while
 296 $N_{eff,EC3}=A_{eff,EC3}f_y$ is the EC3 effective cross-section compression capacity, and $M_{eff,y}=W_{eff,y}f_y$
 297 and $M_{eff,z}=W_{eff,z}f_y$ are the EC3 effective cross-section bending capacities about the major axis
 298 and minor axis, respectively, where $A_{eff,EC3}$ is the EC3 effective cross-section area, and $W_{eff,y}$
 299 and $W_{eff,z}$ are the EC3 effective section moduli about the major axis and minor axis,
 300 respectively.

$$301 \quad \frac{N_{u,pred}}{N_{eff,EC3}} + \frac{M_{u,y}}{M_{eff,y}} + \frac{M_{u,z}}{M_{eff,z}} = 1 \quad (2)$$

302

303 The values of $A_{eff,EC3}$, $W_{eff,y}$ and $W_{eff,z}$ can be determined through using the EC3 effective
 304 width approach. Specifically, $A_{eff,EC3}$ is calculated as the summation of the effective area of
 305 each constituent plate element that is taken as its effective width c_{eff} multiplied by its
 306 thickness t . The effective width c_{eff} can be calculated from Eq. (3), where c is the flat plate
 307 width and $\bar{\lambda}_1$ is the plate slenderness, as given by Eq. (4), in which k_σ is the buckling
 308 parameter and given as 4.0 and 0.43 for internal and outstand plate elements [30], respectively.
 309 With regard to the calculation of $W_{eff,y}$ and $W_{eff,z}$, Eqs (3) and (4) are still used, but the value
 310 of k_σ is different now and dependent on the plate element type (i.e. outstand or internal) and
 311 the stress distribution subjected to bending, as derived based on Table 4.1 and Table 4.2 of EN
 312 1993-1-5 [30].

$$c_{\text{eff}} = \begin{cases} c \left(\frac{1}{\bar{\lambda}_1} - \frac{0.188}{\bar{\lambda}_1^2} \right) \leq c & \text{for outstand elements} \\ c \left(\frac{0.772}{\bar{\lambda}_1} - \frac{0.079}{\bar{\lambda}_1^2} \right) \leq c & \text{for internal elements} \end{cases} \quad (3)$$

$$\bar{\lambda}_1 = \frac{c/t}{28.4 \sqrt{235E/(210000f_y)} \sqrt{k_\sigma}} \quad (4)$$

315

316 Based on the obtained test and FE data, the applicability of the EC3 design interaction curve
 317 to laser-welded stainless steel slender I-sections under combined loading was evaluated.
 318 Graphical evaluation was firstly carried out, with the evaluation results shown in Fig. 10(a)
 319 and Fig. 10(b) for minor-axis and major-axis combined loading cases, where the test and
 320 numerical failure moments and loads, plotted in a normalised format (M_u/M_{eff} versus
 321 $N_u/N_{\text{eff,EC3}}$), are compared against the EC3 linear interaction curve. It was evident that (i) the
 322 EC3 design interaction curve for minor-axis combined loading lies well below the normalised
 323 test and FE data points, indicating excessive conservatism and scatter, and (ii) its major-axis
 324 combined loading counterpart can well represent the corresponding data points. This can be
 325 also proven from the quantitative evaluation results presented in Table 7(a), with the mean test
 326 and numerical to predicted failure load ratios $N_u/N_{u,\text{pred}}$ for minor-axis and major-axis
 327 combined loading cases equal to 1.64 and 1.15, and the corresponding COVs of 0.11 and 0.09.
 328 Given that the EC3 effective compression capacity $N_{\text{eff,EC3}}$ (i.e. the compression end point) is
 329 accurate [8,26], the conservatism and scatter of the EC3 design interaction curve for the
 330 minor-axis combined loading case are due principally to the use of the conservative
 331 minor-axis bending end point (i.e. the effective moment capacity $M_{\text{eff,z}}$) [7,26].

332

333 4.3. ANSI/AISC 370-21 (AISC)

334

335 The design rules given in the American specification ANSI/AISC 370-21 [16] for stainless
 336 steel doubly symmetric cross-sections under combined loading are the same as those set out in
 337 the mild steel design specification ANSI/AISC 360-16 [39]. Specifically, the American
 338 specifications [16,39] adopt a bi-linear interaction curve to predict the cross-section
 339 resistances of slender I-sections under combined compression and bending, as given by Eq.
 340 (5), where $N_{\text{eff,AISC}}=A_{\text{eff,AISC}}f_y$ is the AISC effective cross-section compression capacity and
 341 M_{cy} and M_{cz} are respectively the AISC cross-section effective major-axis and minor-axis
 342 bending capacities, as determined according to Chapter F.5 and F.6 specified in ANSI/AISC
 343 370-21 [16].

$$\begin{cases} \frac{N_{u,\text{pred}}}{N_{\text{eff,AISC}}} + \frac{8}{9} \left(\frac{M_{u,y}}{M_{cy}} + \frac{M_{u,z}}{M_{cz}} \right) = 1.0 & \text{for } \frac{N_{u,\text{pred}}}{N_{\text{eff,AISC}}} \geq 0.2 \\ \frac{N_{u,\text{pred}}}{2N_{\text{eff,AISC}}} + \left(\frac{M_{u,y}}{M_{cy}} + \frac{M_{u,z}}{M_{cz}} \right) = 1.0 & \text{for } \frac{N_{u,\text{pred}}}{N_{\text{eff,AISC}}} < 0.2 \end{cases} \quad (5)$$

345

346 The cross-section resistance predictions for laser-welded stainless steel slender I-sections
 347 under combined loading, as determined from the AISC bi-linear interaction curve, were
 348 quantitatively evaluated against the test and numerical results. Table 7(b) presents the
 349 quantitative evaluation results, where the mean test and numerical to AISC design failure load
 350 ratios $N_u/N_{u,\text{pred}}$ for minor-axis and major-axis combined loading cases are equal to 1.61 and
 351 1.11, respectively. In parallel with the quantitative evaluation, a graphical evaluation was also
 352 performed and shown in Fig. 11, where the experimentally and numerically obtained failure

353 bending moments and axial loads are normalised by the cross-section effective moment
354 capacities and cross-section effective compression capacities, and compared with the AISC
355 bi-linear interaction curve. Both the quantitative and graphical evaluations revealed that (i) the
356 AISC design interaction curve yields overall good design accuracy when applied to
357 laser-welded stainless steel slender I-sections under major-axis combined loading, but with
358 many over-predicted failure load predictions, and (ii) the AISC design interaction curve
359 results in overly conservative and scattered failure load predictions when used for
360 laser-welded stainless steel slender I-sections under minor-axis combined loading, owing
361 mainly to the conservative minor-axis bending end point (M_{cz}).

362

363 *4.4. Continuous strength method (CSM)*

364

365 The continuous strength method (CSM) [17,40] is a recently developed strain-based design
366 method for stainless steel components. For stainless steel slender sections under combined
367 loading, the CSM also adopts the linear interaction curve but with the CSM cross-section
368 bending ($M_{csm,y}$ and $M_{csm,z}$) and compression (N_{csm}) capacities, as given by Eq. (6).

$$369 \quad \frac{N_{u,pred}}{N_{csm}} + \frac{M_{u,y}}{M_{csm,y}} + \frac{M_{u,z}}{M_{csm,z}} = 1 \quad (6)$$

370

371 To calculate the CSM cross-section capacities, the first step lies in determination of the
372 cross-section (compressive) limiting strain ϵ_{csm} that reflects the deformation capacity of the
373 examined slender I-section under the applied loading. This can be achieved by using the ‘base

374 curve' [17], as given by Eq. (7), where $\varepsilon_y=f_y/E$ is the yield strain, $\bar{\lambda}_p = \sqrt{f_y/f_{cr}}$ is the
375 cross-section slenderness, in which f_{cr} is the critical elastic buckling stress of the examined
376 I-section under the applied loading and can be determined by means of the finite-strip
377 package CUFSM [41]. Once the limiting strain ε_{csm} is determined, the CSM design
378 compressive stress σ_{csm} can be calculated as ε_{csm} multiplied by the material Young's modulus
379 E , as given by Eq. (8). Then, the CSM cross-section compression capacity (N_{csm}) and bending
380 capacities ($M_{csm,y}$ and $M_{csm,z}$) are determined by using Eq. (9) and Eq. (10), respectively.

$$381 \quad \frac{\varepsilon_{csm}}{\varepsilon_y} = \left(1 - \frac{0.222}{\bar{\lambda}_p^{1.05}}\right) \frac{1}{\bar{\lambda}_p^{1.05}} \quad (7)$$

$$382 \quad f_{csm} = E\varepsilon_{csm} \quad (8)$$

$$383 \quad N_{csm} = Af_{csm} \quad (9)$$

$$384 \quad \begin{cases} M_{csm,y} = W_{el,y}f_{csm} \\ M_{csm,z} = W_{el,z}f_{csm} \end{cases} \quad (10)$$

385

386 The test and numerical failure moments and loads were normalised by the corresponding M_{csm}
387 and N_{csm} , and are plotted together with the CSM interaction curves in Fig. 12(a) and Fig. 12(b)
388 for minor-axis and major-axis combined loading cases, respectively, where the CSM
389 interaction curve for major-axis combined loading is shown to be capable of well representing
390 the normalised test and FE data while its minor-axis combined loading counterpart lies far
391 below the test and FE data. The graphical evaluation was followed by a quantitative
392 evaluation, with the results given in Table 7(c), where the mean load ratios $N_u/N_{u,pred}$ for
393 minor-axis and major-axis combined loading cases are equal to 1.58 and 1.12. The
394 quantitative and graphical evaluations indicated that similar to the EC3 design interaction

395 curve, the CSM interactive curve provides good design accuracy and consistency for
396 laser-welded stainless steel slender I-sections under major-axis combined loading, but it leads
397 to conservative cross-section resistances predictions for those under minor-axis combined
398 loading, due mainly to the conservative CSM minor-axis bending end point [7,26]. Therefore,
399 new interaction curves that are capable of yielding accurate failure load predications for
400 laser-welded stainless steel slender I-sections under minor-axis combined loading are
401 required.

402

403 *4.5. New design method for minor-axis combined loading*

404

405 The analyses and discussions in Section 4.2 have highlighted that the current EC3 interaction
406 curve for laser-welded stainless steel slender I-sections under minor-axis combined loading
407 has an accurate compression end point but a conservative minor-axis bending end point. An
408 improved design method is therefore proposed by adopting the EC3 interaction curve
409 anchored to a more accurate minor-axis bending end point. As highlighted in previous
410 research [42,43] on the behaviour of steel slender I-sections under minor-axis bending, it is
411 incorrect to assume the linear elastic stress distribution within the slender outstand flanges and
412 plasticity can develop in both compressive and tensile parts. To address the conservatism, the
413 plastic effective width method [42] was developed to consider the plastic reserve capacities of
414 slender steel I-sections in minor-axis bending, and has been proven to yield accurate bending
415 capacity predictions when applied to stainless steel slender I-sections in minor-axis bending
416 [43]. Therefore, the plastic effective width method is ideal to be used to determine the new

417 minor-axis bending end points. The new interaction curve is given by Eq. (11), where $M_{p,eff,z}$
 418 is the effective minor-axis bending capacity determined by the plastic effective width method.

$$419 \quad \frac{N_{u,pred}}{N_{eff,EC3}} + \frac{M_{u,z}}{M_{p,eff,z}} = 1 \quad (11)$$

420

421 In calculation of $M_{p,eff,z}$, the plastic effective width method determines the effective
 422 cross-section based on the strain and stress distributions depicted in Fig. 14 [42,43]. It is
 423 assumed that (i) the maximum attainable compressive strain is equal to the yield strain
 424 multiplied by a coefficient $C_y=3$ and (ii) the plastic compressive region is at a distance of
 425 $e_{cc}=0.225b_f$ from the web centreline, with the region width of b_e determined from Eq. (12).
 426 The location of the neutral axis can then be determined according to the stress equilibrium, as
 427 given by Eq. (13), and the effective minor-axis bending capacity $M_{p,eff,z}$ can be calculated
 428 through integrating the stress distribution, as given by Eqs (14)–(19). Note that the definitions
 429 of the notations in Eqs (13)–(19) are graphically illustrated in Fig. 14.

$$430 \quad b_e = 0.2b_f \bar{\lambda}_p^{-0.75} \quad (12)$$

$$431 \quad x_p = \frac{2b_e t_f [(b_f - b_e / 2) - (0.5b_f - b_e - e_{cc})] + b_f t_f b_f / 4 + (h - 2t_f) t_w b_f / 2}{2b_e t_f + b_f t_f + (h - 2t_f) t_w} \quad (13)$$

$$432 \quad M_{p,eff,z} = 2b_e t_f f_y \left(e_{cc} + \frac{b_e}{2} + c \right) + 2b_p t_f f_y \left(x_p - \frac{b_p}{2} \right) + \frac{2}{3} b_g^2 f_y t_f + \frac{2}{3} c^2 f_w t_f + (h - 2t_f) t_w f_w c \quad (14)$$

$$433 \quad \text{where } b_p = x_p - b_g \quad (15)$$

$$434 \quad b_g = \varepsilon_y / K \quad (16)$$

$$435 \quad K = \frac{C_y \varepsilon_y}{0.5b_f - x_p + e_{cc} + b_e} \quad (17)$$

$$436 \quad c = 0.5b_f - b_g - b_p \quad (18)$$

437
$$f_w = cKE \tag{19}$$

438

439 The normalised test and numerical failure moments ($M_u/M_{peff,z}$) and loads ($N_u/N_{eff,EC3}$) are
440 plotted against the new design interaction curve in Fig. 13, where the new interaction curve
441 displays a much better representation of the normalised data points in comparison with its
442 EC3, AISC and CSM counterparts in Figs 10–12, revealing the substantially improved design
443 accuracy and consistency. This is also evident in Table 7(d), where the mean load ratio
444 $N_u/N_{u,pred}$ is equal to 1.21, with the corresponding COV of 0.07, indicating that the proposed
445 new interaction curve leads to significantly more accurate and consistent failure load
446 predictions for laser-welded stainless steel slender I-sections under minor-axis combined
447 loading than the three considered methods [15–17].

448

449 The reliability of the new design interaction curve when applied to laser-welded stainless steel
450 slender I-sections under minor-axis combined loading was assessed herein, according to the
451 requirements and procedures given in EN 1990 [44]. In the present reliability analysis, the
452 material over-strength ratio for austenitic stainless steel and the corresponding COV were
453 taken as 1.3 and 0.06, respectively, and the COV of the geometric properties of stainless steel
454 cross-sections was taken as 0.05, following the recommendations of Afshan et al. [45]. The
455 key statistical parameters calculated according to EN 1990 [44] are presented in Table 8,
456 where $k_{d,n}$ is the design (ultimate limit state) fractile factor, b is the mean ratio of the test and
457 numerical to design model resistances, V_δ is the COV of the test and numerical resistances
458 relative to the resistance model, V_r is the combined COV incorporating both model and basic

459 variable uncertainties, and γ_{M1} is the partial safety factor. The resulting (required) partial
460 safety factor for the EC3 design rules, as reported in Table 8, is equal to 0.99, less than the
461 current used value of 1.1 in EN 1993-1-4 [15], therefore demonstrating the reliability of the
462 new design method when applied to laser-welded stainless steel slender I-sections under
463 minor-axis combined loading.

464

465 **5. Conclusions**

466

467 A thorough testing and numerical modelling programme has been performed to investigate the
468 cross-section resistances and local stability of laser-welded stainless steel slender I-sections
469 under combined compression and bending. A testing programme was firstly conducted, which
470 included initial local geometric imperfection measurements and eccentric compression
471 experiments on ten laser-welded stainless steel slender I-section stub column specimens, with
472 the test procedures and results fully presented. The testing programme was supplemented by a
473 numerical simulation programme, in which FE models were developed and validated against
474 the test observations. Upon validation, the FE models were employed to perform parametric
475 studies to generate a numerical data pool over a broad range of cross-section dimensions and
476 loading combinations. Then, the obtained test and numerical data were used to graphically
477 and quantitatively evaluate the applicability of the design interaction curves given in EN
478 1993-1-4 [15], ANSI/AISC 370-21 [16] and the CSM [17] for laser-welded stainless steel
479 slender I-sections under combined loading. The evaluation results generally indicated that all
480 the considered EC3, AISC and CSM interaction curves offer good design accuracy for

481 laser-welded stainless steel slender I-sections under major-axis combined loading, but lead to
482 overly conservative and scattered cross-section resistance predictions for their counterparts
483 under minor-axis combined loading, owing principally to unduly conservative minor-axis
484 bending end points. To address this shortcoming, an improved design interaction curve was
485 proposed for laser-welded stainless steel slender I-sections under minor-axis combined
486 loading, which adopts the EC3 interaction curve anchored to a more accurate minor-axis
487 bending end point. The proposed interaction curve was shown to offer substantially improved
488 design accuracy and consistency than the three considered curves. A reliability analysis was
489 then performed to confirm the reliability of the proposed design interaction curve.

490

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492

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496

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601 **Figures**

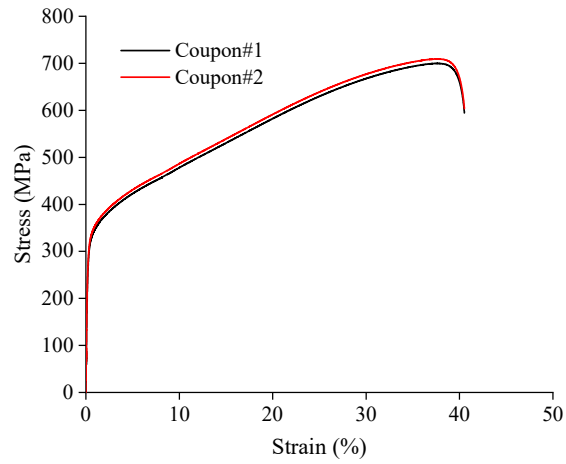


Fig. 1. Measured stress–strain curves.

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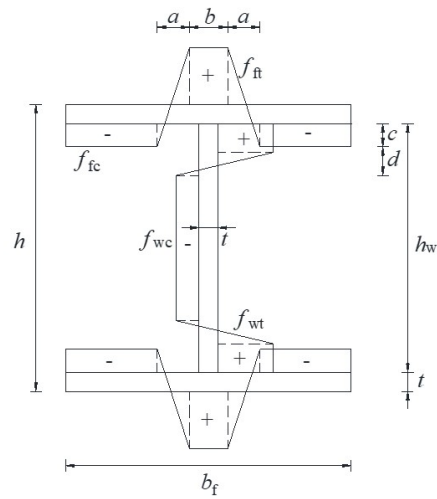


Fig. 2. Membrane residual stress predictive model for laser-welded stainless steel I-sections (+ve = tension; -ve = compression).

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Fig. 3. Setup for initial local geometric imperfection measurements.

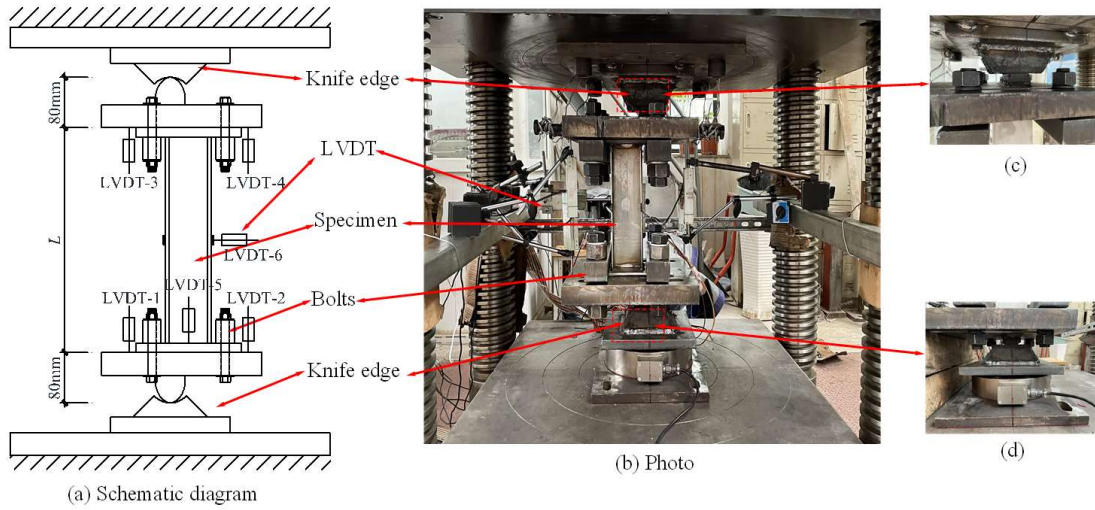


Fig. 4. Eccentric compression test setup.

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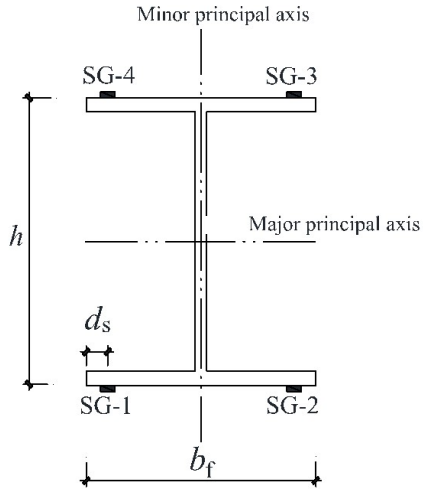


Fig. 5. Detailed positions of strain gauges.

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(a) Minor-axis combined loading

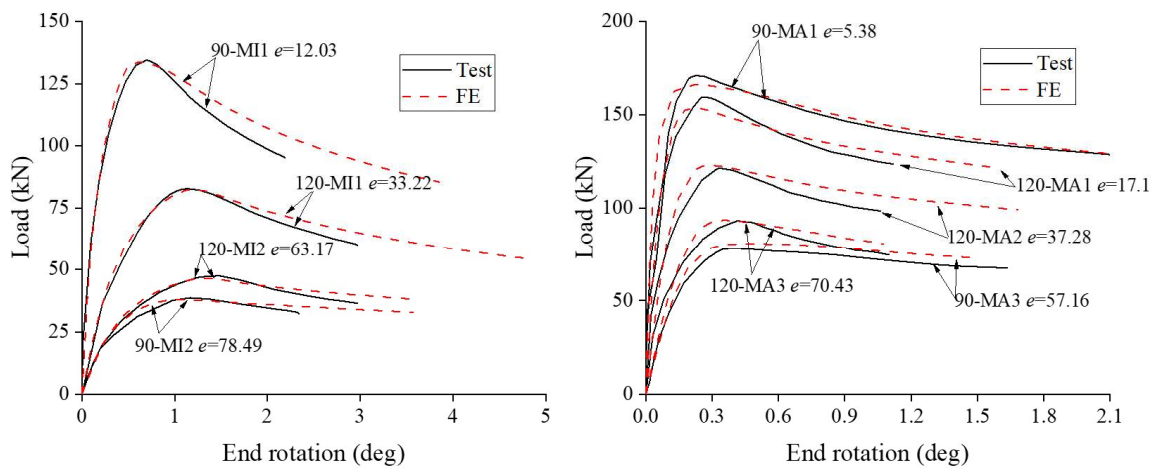


(b) Major-axis combined loading

Fig. 6. Failure modes of laser-welded stainless steel slender I-section stub column specimens under combined loading.

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(a) Minor-axis combined loading

(b) Major-axis combined loading

Fig. 7. Load–end rotation curves of laser-welded stainless steel slender I-section stub column specimens under combined loading.

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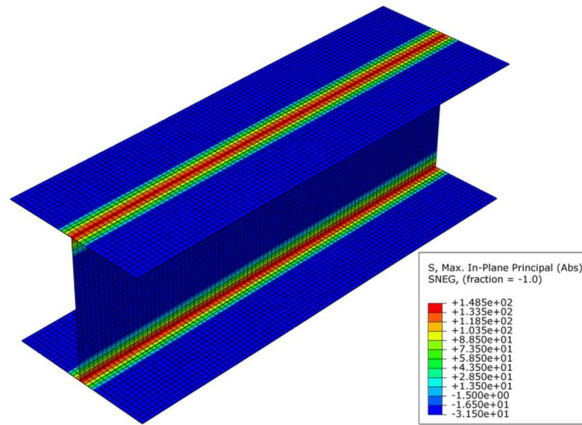
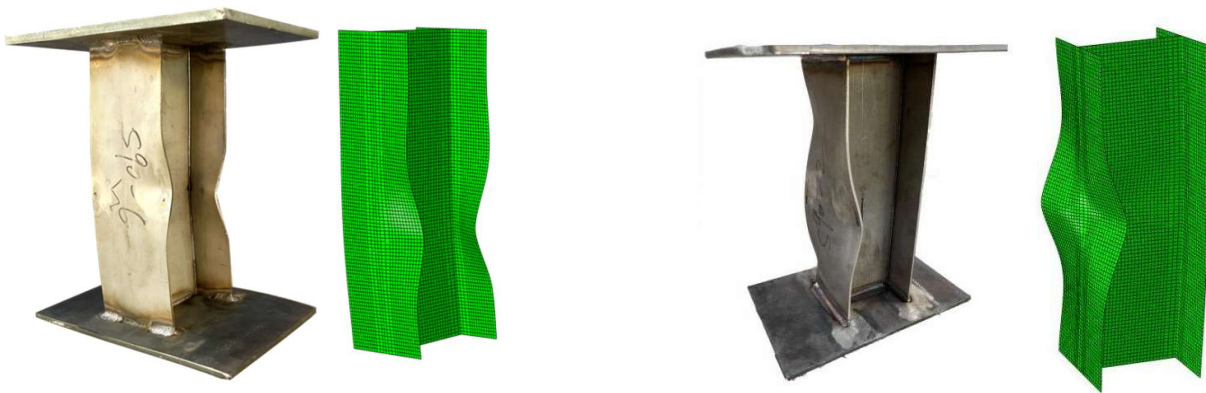


Fig. 8. Typical membrane residual stresses (in MPa) for modelled I-90×90×3 specimens.

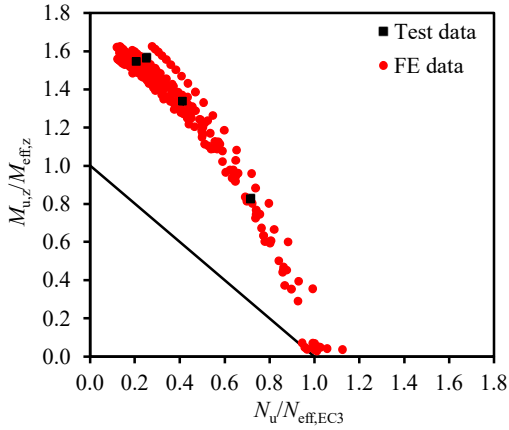
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(a) Specimen 90-MI2 (minor-axis combined loading) (b) Specimen 90-MA3 (major-axis combined loading)

Fig. 9. Test and FE failure modes of typical specimens under combined loading.

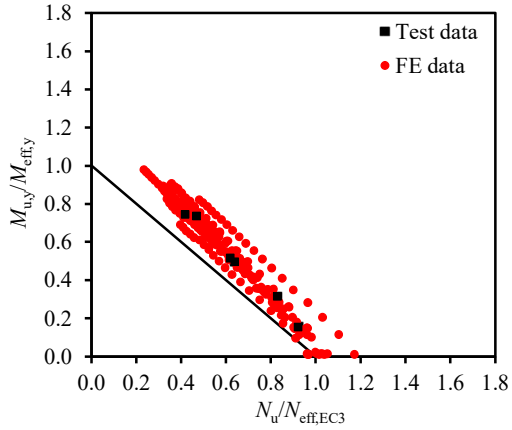
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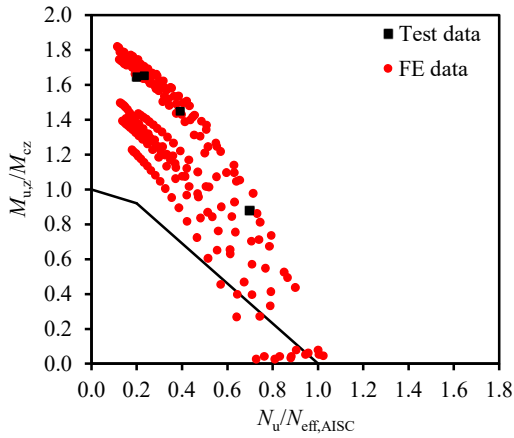
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(a) Minor-axis combined loading



(b) Major-axis combined loading

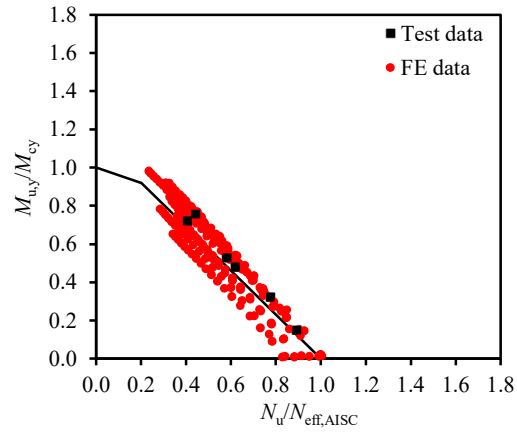
Fig. 10. Comparisons of test and numerical failure loads with EC3 predicted failure loads.



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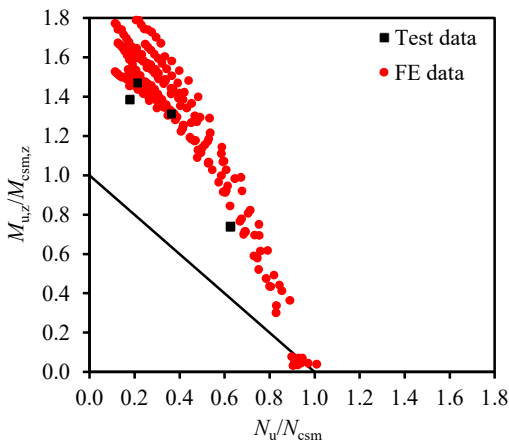
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(a) Minor-axis combined loading



(b) Major-axis combined loading

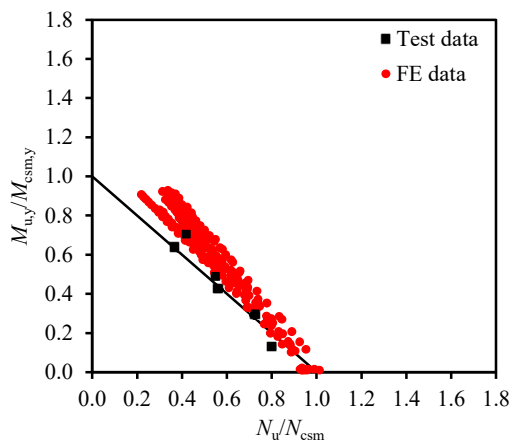
Fig. 11. Comparisons of test and numerical failure loads with AISC predicted failure loads.



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(a) Minor-axis combined loading



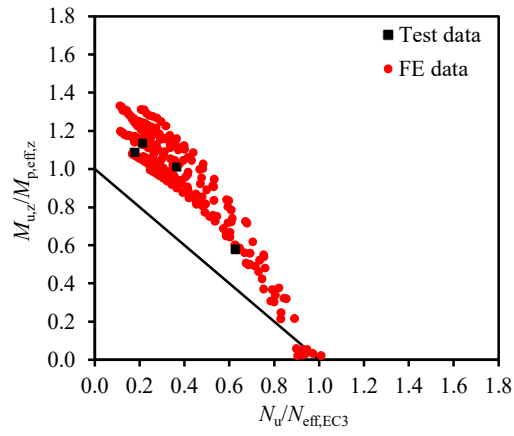
(b) Major-axis combined loading

Fig. 12. Comparisons of test and numerical failure loads with CSM predicted failure loads.

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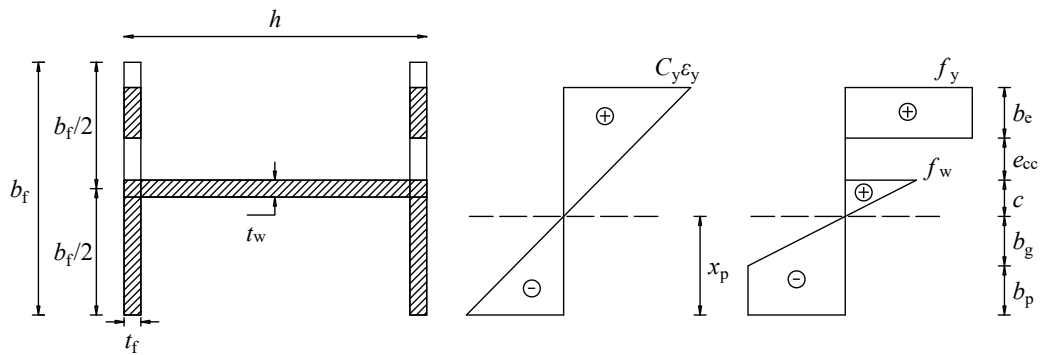
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Fig. 13. Comparisons of test and numerical failure loads with new proposal predicted failure loads.

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Fig. 14. Strain and stress distributions in slender flanges.

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650 **Tables**

651

652

Table 1

Key average measured material properties.

E (MPa)	f_y (MPa)	$f_{1.0}$ (MPa)	f_u (MPa)	ϵ_u (%)	ϵ_f (%)	R–O coefficients	
						n	$m_{1.0}$
189000	297	349	704	38	60	5.3	2.9

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Table 2

Membrane residual stress predictive model for laser-welded stainless steel I-sections [6].

$f_{\bar{n}} = f_{wt}$ (tension)	$f_{\bar{c}} = f_{wc}$ (compression)	a	b	c	d
$0.5f_y$	From equilibrium	$0.1b_f$	$0.075b_f$	$0.0375h_w$	$0.1h_w$

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Table 3

Geometric properties of laser-welded stainless steel slender I-section stub column specimens under combined loading.

Axis of combined loading	Cross-section	Specimen ID	L (mm)	h (mm)	b_f (mm)	t (mm)	ω_0 (mm)	ω_0/b_f
Minor axis	I-90×90×3	90-MI1	270.1	89.22	90.00	2.76	0.30	1/300
	I-90×90×3	90-MI2	269.2	89.18	89.97	2.75	0.23	1/391
	I-120×90×3	120-MI1	360.1	119.40	89.97	2.81	0.25	1/360
	I-120×90×3	120-MI2	361.2	119.48	90.08	2.72	0.28	1/322
Major axis	I-90×90×3	90-MA1	270.5	89.39	90.05	2.74	0.35	1/257
	I-90×90×3	90-MA2	270.0	89.51	90.01	2.75	0.26	1/346
	I-90×90×3	90-MA3	270.0	89.27	89.95	2.75	0.26	1/345
	I-120×90×3	120-MA1	359.8	119.51	89.94	2.76	0.31	1/290
	I-120×90×3	120-MA2	360.0	119.27	89.93	2.79	0.30	1/300
	I-120×90×3	120-MA3	360.0	119.20	89.91	2.81	0.28	1/321

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Table 4

Key test results for laser-welded stainless steel slender I-section specimens under combined loading.

Axis of combined loading	Specimen ID	e_0 (mm)	N_u (kN)	Δ_u (mm)	M_u (kNm)	δ_u (mm)	ϕ_u (deg)
Minor axis	90-MI1	12.03	134.53	3.23	2.05	0.85	0.69
	90-MI2	78.49	38.61	4.46	3.20	3.36	1.09
	120-MI1	33.22	82.76	3.87	3.07	1.68	1.15
	120-MI2	63.16	47.47	4.12	3.19	3.22	1.49
Major axis	90-MA1	5.38	170.93	0.31	0.97	1.03	0.23
	90-MA2	25.04	119.47	1.58	3.18	– *	– *
	90-MA3	57.16	78.13	2.58	4.67	1.89	0.46
	120-MA1	17.17	159.49	1.11	2.92	0.97	0.29
	120-MA2	37.28	121.42	2.21	4.79	1.41	0.40
	120-MA3	70.43	93.09	3.32	6.86	1.83	0.46

* LVDTs were disconnected to data logger during testing – LVDT readings could not be obtained.

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Table 5

Comparison between test and numerical failure loads for measured and generalised local imperfection magnitudes.

Axis of combined loading	Specimen ID	FE N_u / Test N_u	
		ω_0	$b/300$
Minor axis	90-MI1	1.00	1.01
	90-MI2	1.02	1.02
	120-MI1	1.00	1.00
	120-MI2	1.02	1.01
Major axis	90-MA1	1.02	1.03
	90-MA2	0.98	0.98
	90-MA3	1.00	1.00
	120-MA1	1.04	1.04
	120-MA2	0.99	0.99
	120-MA3	0.96	0.96
	Mean	1.01	1.01
COV	0.02	0.02	

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Table 6

Cross-section dimensions and loading eccentricities of modelled laser-welded stainless steel slender I-sections in parametric studies.

h (mm)	b (mm)	t (mm)	e_0 (mm)
60, 90, 120, 150	60, 90, 120, 150	2, 2.2, 2.4, 2.6, 2.8, 3	1, 5, 10, 15, 20, 25, 30, 35, 40, 45, 50, 55, 60, 65, 70, 75, 80, 85, 90, 95, 100

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Table 7

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Comparisons of test and numerical failure loads with predicted failure loads.

(a) EN 1993-1-4 [15]

Loading condition	No. of test data	No. of FE data	$N_u/N_{u,pred}$	
			Mean	COV
Minor-axis combined loading	4	200	1.64	0.11
Major-axis combined loading	6	200	1.15	0.09

(b) AISI/AISC 370-21 [16]

Loading condition	No. of test data	No. of FE data	$N_u/N_{u,pred}$	
			Mean	COV
Minor-axis combined loading	4	200	1.61	0.16
Major-axis combined loading	6	200	1.11	0.08

(c) CSM [17]

Loading condition	No. of test data	No. of FE data	$N_u/N_{u,pred}$	
			Mean	COV
Minor-axis combined loading	4	200	1.58	0.11
Major-axis combined loading	6	200	1.12	0.07

(d) New proposal

Loading condition	No. of test data	No. of FE data	$N_u/N_{u,pred}$	
			Mean	COV
Minor-axis combined loading	4	200	1.21	0.07

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Table 8

Reliability analysis results for new design method according to EN 1990 [44].

No. of test and FE data	$k_{d,n}$	b	V_δ	V_R	γ_{M1}
204	3.14	1.17	0.11	0.13	0.99

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